

# INTEGRATED SYSTEMS PERFORMANCE ASSESSMENT FOR THE EVALUATION OF SPACE NUCLEAR REACTOR DESIGN CONCEPTS (PHASE 1: DEMONSTRATION OF THE METHODOLOGY)

Steven D. Clement  
Christopher N. Amos

Science Applications International Corporation  
2109 Air Park Rd. SE  
Albuquerque, NM 87106

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Final Report

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**AIR FORCE MATERIEL COMMAND**  
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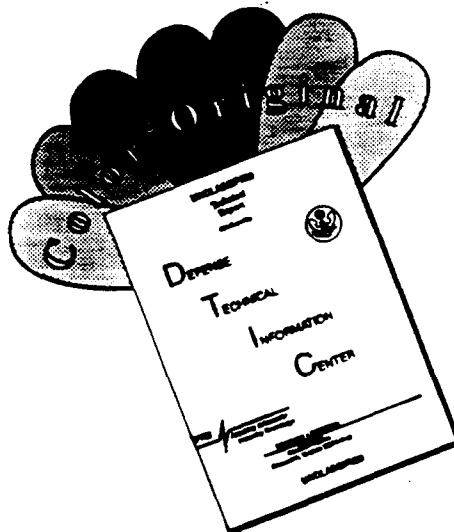
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DOUG DELL, Capt, USAF  
Project Officer

FOR THE COMMANDER



DAVID KRISTENSEN, Lt Col, USAF  
Chief, Space Power and Thermal  
Management Division



HENRY L. PUGH, JR., Col, USAF  
Director of Space and Missiles Technology

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## Acronym List

DOE	U.S. Department of Energy
ISPA	Integrated Systems Performance Assessment
PL	Phillips Laboratory
PSA	Probabilistic Safety Assessment
QFD	Quality Functional Deployment
SAIC	Science Applications International Corporation
SNL	Sandia National Laboratories
SNPS	Space Nuclear Power System
TFE	Thermionic Fuel Element
TQM	Total Quality Management
USAF	U.S. Air Force

## 1.0 INTRODUCTION

The U.S. Air Force (USAF) Phillips Laboratory (PL) has identified five basic research areas in Space Power and Thermal Management where the development of innovative technologies has been requested. These technologies are anticipated to provide significant payoff to the performance, survivability, and affordability of important systems. The first basic research area is Space Nuclear Power. This report discusses the development and application of an innovative approach to systems performance assessment for the evaluation of space nuclear reactor design concepts. The approach described is anticipated to provide an important tool for guiding the assessment of component and system performance. The extension of this approach to provide a framework for design optimization is also envisioned.

### 1.1 Space Nuclear Power Systems

PL's objectives for research in the area of space nuclear power are directed towards experimental and analytical development of advanced space nuclear power systems (SNPSs) for future USAF spacecraft. Specifically, PL's focus is on advancing the state-of-the-art technology in SNPS materials, components, subsystems, and systems for power generation, conversion, and distribution, as well as heat transport/rejection for existing and proposed system concepts. To achieve these objectives, PL has identified the following specific areas for research:

- Increase thermal to electric conversion efficiency
- Reduce production costs
- Increase lifetime
- Reduce system mass
- Increase reliability
- Identify and analyze mission applications and their attendant requirements
- Develop improved numerical and analytical techniques for designing and assessing space power systems
- Reduce development cost
- Reduce testing risk, cost, and schedule

Each of these research areas involves independent assessment and subsequent integration of individual components or subsystems into an SNPS. The proposed system designs are largely

conceptual, and many of the proposed materials, physical processes, and components have limited modeling, testing, and validation on which to base the assessment of their performance within the overall system. This lack of information relative to certain aspects of conceptual designs gives rise to uncertainty relative to the overall performance of the system. Therefore, a potentially high degree of uncertainty can exist with regard to ultimate performance (i.e., power output and capacity), mission application, reliability, lifetime, development cost, and testing risk. This uncertainty has been termed "technology risk."

PL has identified two specific space nuclear reactor design concepts to be evaluated using Integrated Systems Performance Assessment (ISPA) methodology: the S-PRIME concept designed by Rockwell International and the SPACE-R concept designed by Space Power Incorporated. Each design is characterized by a moderated core, a NaK pumped loop primary coolant system, and a potassium heat pipe radiator as the ultimate heat sink. The most significant difference between the two concepts is in the design of the in-core Thermionic Fuel Elements (TFEs). The S-PRIME concept proposes the use of multi-cell TFEs, while the SPACE-R concept proposes the use of core-length single-cell TFEs.

The TFE design and performance specifications have a significant impact on the assessment of overall system performance. The TFE design could affect the system cost, demonstrable reliability, scalability, power, etc. The impact on the overall system performance could be either positive or negative, depending on the evaluation criteria being considered. This is an example of only one component of a complex system. A fully integrated engineered system could potentially have hundreds of components, each of whose individual performance is dependent on the performance of others. To effectively evaluate overall system performance, an integrated model that considers both multiple levels of dependencies and weighted evaluation criteria is required.

## **1.2 Integrated Systems Performance Assessment**

SAIC has coined the term ISPA to refer to the application of an inductive probabilistic modeling framework to the assessment of the overall performance of engineered systems. This framework provides PL with the appropriate tool for the application of existing models for physical processes and component performance to an integral assessment of SNPSs that enables the assessment of performance combined with a consistent evaluation of technology risk.

Inductive modeling begins by characterizing the system performance in terms of acceptance criteria (e.g., maximum power output, mass, and reliability) and/or its response to events (e.g., failure of a component or exposure to excessive heat). The analyst then describes the causal relationships between these performance criteria and relevant events. The system is then designed based on inductive reasoning. Event trees have been used in reliability engineering and safety assessments to represent these causal relationships. Specifically, the analyst indicates, using the event-tree structure, the sequence of design features, physical processes, and component performance required to attain a certain performance goal, either with respect to each acceptance criterion or in response to an event. The structure of the tree indicates the causal relationships that the analyst believes exist. Because the event-tree modeling framework is probabilistic, the analyst can indicate, within the context of the model, his or her confidence that a particular causal relationship is accurate. Multiple relationships with varying levels of confidence can be indicated if the analyst identifies more than one potential cause-effect relationship. The event-tree framework provides a rigorous, traceable method for performing the evaluation that allows a fully consistent parallel assessment of technology risk.

The EVNTRE<sup>1</sup> code, developed by Sandia National Laboratories (SNL) for use in nuclear plant safety assessments, is used to evaluate the event tree models for the performance of each conceptual design. EVNTRE is extremely powerful in this regard because it allows the inclusion of any user-supplied code to evaluate any branch point (i.e., performance indicator). It also allows the use of parameters (e.g., temperatures, flow rates, and heat fluxes) within the framework to characterize the interactions and dependencies between models.

### 1.3 Report Organization

This report describes the development of an ISPA model for the evaluation of two space nuclear reactor design concepts. Section 1.0 (this section) introduces the problem of evaluating complex engineered systems and the ISPA methodology. Section 2.0 describes the ISPA methodology. Section 3.0 describes the Phase I ISPA model development, including a description of the evaluation criteria and the parameters used to assess the success or failure to achieve specific design goals. Section 4.0 discusses the results of the assessment and identifies how key uncertainties and sensitivities may be identified. Finally, Sections 5.0 and 6.0 provide a summary of the results and outline the tasks associated with Phases II and III of the ISPA project.

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1 Greismeyer, J. M., and L. N. Smith, *A Reference Manual for the Event Progression Analysis Code (EVNTRE)*, NUREG/CR-5174.

## 2.0 EVALUATION OF SPACE NUCLEAR REACTORS

Evaluation of complex engineered system performance with respect to a given set of criteria requires an integrated assessment that considers physical processes, component behavior, and constraints imposed by the intended application (e.g., size and mission time). Dependencies between these considerations lead to the requirement for an integrated model. Since some models already exist for assessing component behavior and the associated physical processes, a framework that allows integration of individual models is particularly appealing. The ISPA methodology provides such a framework. Furthermore, since uncertainty exists with respect to both the appropriateness (or correctness) of models for some physical processes and the performance of components under certain conditions, the ability to reflect that uncertainty within the integrated assessment provides a particularly powerful tool. Because ISPA is based on probabilistic modeling methodology, it enables the simultaneous modeling of system performance (based on process and component performance models) and assessment of technology risk (i.e., the confidence or likelihood of attaining the level of performance indicated by a particular combination of models).

### 2.1 Integrated Analysis Requirements

The focus of analysis efforts with regard to commercial nuclear power plants, and more recently the U.S. Department of Energy's (DOE's) nuclear material production facilities, has been safety assessment. It should be recognized, however, that safety assessment is the assessment of reactor system performance in response to anticipated events and equipment failures. The methods that have been developed are thus equally applicable, and perhaps of greater overall value, in the assessment of system performance relative to established criteria other than safety. Such criteria may include power produced, reliability, survivability, and others related to suitability for the intended mission.

Experience derived from more than two decades of effort to characterize the safety performance of commercial nuclear plants suggests that such efforts must consider the whole system. Efforts that have considered each component individually, or based on a limited characterization of its interaction with the other system components, have generally been shown to be less than adequate. It is frequently the case that a seemingly minor aspect of the design or operation of one component can have a significant effect on the performance of another. Inductive modeling techniques that focus the analyst's attention on cause and effect relationships within the system have become an

important tool in the assessment of existing nuclear reactor systems. These types of analyses draw on results obtained from integrated mechanistic tools, component models, detailed analysis of specific phenomena, and test results to develop a more complete picture of system performance.

Nuclear power systems have been studied for more than three decades. Even with the large body of performance data and models that have resulted, significant areas remain where knowledge is less than complete. As a consequence, significant performance uncertainties remain. The probabilistic modeling framework of the ISPA methodology allows treatment of these uncertainties in a manner that permits the analyst to evaluate their potential effect on system performance. Of even greater benefit is the fact that system models constructed within this framework can be readily modified to evaluate design or operational adjustments that minimize the potential effect of these uncertainties on system performance. Techniques have also been developed to allow the analyst to identify which uncertainties included in the model have the greatest impact on the uncertainty in system performance. This information can be used to focus model development and component testing so as to gain the greatest benefit in minimizing technology risk.

## **2.2 Description of the ISPA Methodology**

ISPA is a term that has been recently adopted by Science Applications International Corporation (SAIC) to describe a methodology that we have adapted from the nuclear plant risk assessment field for more general application to the assessment of engineered systems. The ISPA methodology has been designed to address the needs that currently exist in several technological areas with respect to developing models for the performance of complex engineered systems. ISPA provides a framework within which the analyst can develop a model for the dependencies and interactions between components, phenomena, and human or computer controls. That framework allows the analyst to develop a system model by importing models for physical phenomena (e.g., heat transfer correlations), component performance (e.g., heat pipe performance code), and component reliability (e.g., fault trees), and combining them based on identified interactions and dependencies. ISPA also allows for assessment of system performance uncertainty that may be inherent due to uncertainties related to the appropriateness or accuracy of the imported models.

This section provides a brief background of the methodology, discusses the specific benefits of this approach for the performance assessment of space reactor design concepts, and describes the modeling approach.

## *History*

ISPA is based on the application of software developed by SNL to support probabilistic assessments of the safety of nuclear power plants. The software consists of two computer codes: EVNTRE, the principal code, and PSTEVNT, which is a utility code that facilitates the analysis of EVNTRE results. Both codes are documented in the open literature and are available to all domestic (U.S.) concerns.

EVNTRE provides a framework for the development of inductive models for system performance. Its structure allows relationships between physical phenomena, component performance, and human or control system interaction to be expressed in a completely general fashion. Boolean expressions are used within the framework to express these relationships in a manner similar to that used in conventional reliability analysis software. EVNTRE represents a significant advance over reliability software in that it allows the inclusion of mechanistic models to evaluate any aspect of system performance that may be required. In existing applications, these models are typically correlations or tabulations of the results obtained from more sophisticated computer models. However, incorporation of detailed mechanistic models is possible.

Inductive modeling begins with the identification of performance criteria or goals. Initially, deductive reasoning is used to identify the component performance measures that allow evaluation of performance relative to these goals or criteria. This process can be described as the analyst asking which components must function at what level to achieve the desired performance. The inductive process is then applied to include consideration of system effects. This part of the modeling process involves asking questions about the effect of an anticipated or potential behavior of one subsystem or component on the behavior of another. These questions, and the methods used to evaluate the answers, constitute the event tree model that the EVNTRE code will then evaluate. The framework is sufficiently flexible to allow easy enhancement by adding additional questions, including newly recognized dependencies, and enhancing the mechanistic models used to evaluate physical processes or component behavior.

This approach to system performance modeling parallels decision analysis techniques that have been developed to assist in technical management. The event tree that is formed by the series of questions posed by the analyst defines the context for each subsequent answer. The set of answers to previous questions in the analysis thus defines the context in which each question is answered. Rules for the evaluation of each question based on this context are then defined. Rule-dictated evaluation is the key element in decision analysis and is the core of the proposed methodology. This

methodology, which combines mechanistic component modeling, probabilistic assessment, and decision analysis into an integrated performance model, is ideally suited for the performance assessment of complex conceptual designs.

### *Description*

The ISPA framework is best represented by an event tree. Event trees have been used extensively, and successfully, to evaluate system performance and response to off-normal events. They can reflect interdependencies and treat significant uncertainties. The ISPA methodology is a logical extension of Probabilistic Safety Assessment (PSA). The difference between the two is that performance measures other than accident consequences are the desired results. The ISPA model developed for PL is an event tree with various SNPS performance criteria as top events. Each top event is characterized in terms of discrete levels (e.g., high, medium, or low) of performance. Each level corresponds to a specific range in the value of a parameter that characterizes performance for a specific criterion and is represented by a branch in the tree. Evaluating (or quantifying) the tree requires evaluating the parameter values corresponding to each level and assigning a split fraction (i.e., confidence that the outcome indicated by that branch is correct) to that branch. By multiplying the branch split fractions together along a pathway through the tree (typically referred to as a sequence), the frequency of each path through the tree (which indicates a certain level of performance relative to each criterion) is evaluated.

This approach offers significant advantages in the way the split fractions are evaluated relative to conventional event tree tools. Determination of the importance or significance of uncertainties in the performance of a system or component is facilitated. The top event split fractions can be evaluated independently, as a function of the outcome of previous events, or based on the results of an external user-defined function. Dependencies between systems can be handled within the model through the use of Boolean expressions that define the relationship. Also, independent component models can be linked with the tree for a detailed, systematic evaluation of key parameters that describe component performance. Multiple user functions that perform this type of evaluation are possible, facilitating the assessment of individual systems with various models. Evaluation of the sensitivity of the prediction of system performance to the various models can be evaluated as an integral part of the assessment.

Final evaluation of a particular space nuclear reactor design concept is performed by "scoring" each individual sequence. Each performance indicator will be assigned a weighting factor representing the importance of the characteristic to mission success. For example, sequences that indicate



demonstrable reliability are assigned a higher weight than sequences characterized by no demonstrable reliability. The score of a particular sequence is the sum of its performance indicator weighting factors. The magnitude of the score is representative of the probability that a particular sequence will achieve the desired result. Multiplying each sequence score by its calculated frequency and summing over all sequences results in an overall score or an evaluation of the design concept. This method combines measures for system performance and technology risk. Other approaches to producing an overall score are also possible.

### ***Benefits of ISPA***

Using the ISPA methodology provides an exceptional tool for evaluating the effect of uncertainties in individual systems to the integrated system performance. ISPA models can be modified and exercised quickly and inexpensively. By modifying uncertain parameters and quantifying the model, the analyst can determine the frequency change of selected performance evaluation criteria. The effect of individual design parameters, components, or features on the overall performance of the integrated system can therefore be assessed. This process provides the necessary information to increase the overall system reliability, increase the lifetime, and analyze specific mission applications, and it provides a basis for reducing development cost and assessing risk. Specific advantages include the following:

- The methodology encourages careful consideration of component dependencies and interactions that have been shown to have significant implications for nuclear reactor system performance.
- Existing models for component performance can be incorporated in the model, either directly or by summarizing the results.
- Technology risk and system performance are assessed simultaneously, providing greater assurance of consistency.
- The evaluation is traceable and reproducible since it is based on an integrated system model rather than individual evaluations against diverse, and sometimes subjective, performance criteria.
- The resulting system model is easily modified providing significant flexibility in reflecting new insights or adding more detail to the performance evaluation.
- Once a system model has been developed, the evaluation of design tradeoffs, the effectiveness of testing in reducing technology risk, and the enhancement of the model to enable more detailed predictions of system performance is facilitated.

### **3.0 PHASE I ISPA MODEL DEVELOPMENT**

Phase I of the PL ISPA project consists of the development of a simple model for the overall system performance assessment of a space nuclear reactor design concept. The goal of Phase I is to demonstrate the ISPA methodology and its potential as an integrated assessment tool. The following sections describe the selection of evaluation criteria; the identification of individual systems or components to be included in the model; and, finally, their integration into an ISPA model.

#### **3.1 Evaluation of Space Nuclear Power Systems**

As described earlier in this document, the first step in developing a model for systems performance assessment is to establish the evaluation criteria. These criteria provide the quantitative basis for the performance assessment.

##### **3.1.1 Quality Functional Deployment (QFD) Analysis**

As part of a Total Quality Management (TQM) exercise, PL has initiated a Quality Functional Deployment (QFD) process to assist in identifying important characteristics of an SNPS. The QFD process consisted of establishing a list of "wants" in a thermionic space power system (through a brain-storming exercise) and ranking those "wants" by systematically evaluating (through expert elicitations) their importance. The results of the QFD process provide two key pieces of information required for the ISPA model: (1) the list of "wants" in a thermionic SNPS establishes the evaluation criteria, and (2) the ranking of those "wants" provides their relative weight or importance. Table 1 lists the results of the brain-storming exercise (i.e., the "wants") and groups them into eleven primary evaluation criteria categories. Table 2 lists the eleven primary evaluation criteria and their relative importance based on the ranking each received.

**Table 1. "Wants" in a Thermionic Space Power System**

<b>Evaluation Criteria Group<sup>a</sup></b>	<b>"Wants"<sup>b</sup></b>
<b><i>Attractive Costs</i></b>	<ol style="list-style-type: none"> <li>1. Low up front development costs</li> <li>2. Affordable demonstration costs (fabrication, qualification, and first flight)</li> <li>3. Low life cycle costs</li> </ol>
<b><i>Flexible Operation</i></b>	<ol style="list-style-type: none"> <li>4. Start up and forget</li> <li>5. Off design power operation (short term; e.g., days)</li> <li>6. Multiple startups and shutdowns on orbit</li> <li>7. Operate in a variety of environments or orbits, such as the surface of the Moon and Mars, without changing design (costs acceptable; parasitic vacuum vessel not allowed)</li> </ol>
<b><i>Easy to Build, Handle, and Maintain</i></b>	<ol style="list-style-type: none"> <li>8. Easy to fabricate</li> <li>9. Transportable, easy to move</li> <li>10. Repairable until launch</li> <li>11. Repairable on orbit</li> </ol>
<b><i>Compatible with Satellite/ Mission Objectives</i></b>	<ol style="list-style-type: none"> <li>12. Minimal negative impact on satellite</li> <li>13. Improve operational performance of satellite</li> <li>14. Allow use of smaller launch vehicle</li> <li>15. Ability to support secondary function</li> <li>16. Standard interface for coupling to satellite (e.g., electrical and mechanical)</li> </ol>
<b><i>Enhance Survivability of Satellite</i></b>	<ol style="list-style-type: none"> <li>17. Avoid detection (i.e., hideability—small signature design)</li> <li>18. Avoid attack (i.e., ability to withstand maneuvering)</li> <li>19. Survive attack</li> <li>20. Operate through attack</li> </ol>
<b><i>Reliable Design</i></b>	<ol style="list-style-type: none"> <li>21. Demonstrate reliability</li> <li>22. No credible mission-ending single point failures</li> </ol>
<b><i>Flexible/Scalable Design</i></b>	<ol style="list-style-type: none"> <li>23. Scalable over small power range with no design changes</li> <li>24. Scalable over medium power range with no technology or component design changes</li> <li>25. Scalable technology over large power range</li> <li>26. Simplicity—use common components within same reactor</li> <li>27. Simplicity—use common components while scaling power</li> <li>28. Ability to easily incorporate technological advancements</li> </ol>
<b><i>Safety and Public Acceptance</i></b>	<ol style="list-style-type: none"> <li>29. Low radiological risk to biosphere</li> <li>30. Saleable to public (safety, cost, and benefit)</li> <li>31. Man-rated</li> </ol>
<b><i>Schedule</i></b>	<ol style="list-style-type: none"> <li>32. Key technologies demonstrated by 10/94</li> <li>33. Fly by 10/99</li> </ol>
<b><i>Acceptable Program Risk</i></b>	<ol style="list-style-type: none"> <li>34. Acceptable development risk (high probability that development will be successful—technical, cost, schedule)</li> <li>35. Performance must be assessable when satellite is in orbit</li> <li>36. Testable—full system qualification and acceptance</li> </ol>
<b><i>Attractive Lifetime</i></b>	<ol style="list-style-type: none"> <li>37. Long operational life</li> <li>38. Lifetime tailorable to mission</li> </ol>

<sup>a</sup> Also referred to as Level 1 evaluation criteria

<sup>b</sup> Also referred to as Level 2 evaluation criteria

**Table 2. Ranking of Space Nuclear Power System ISPA Evaluation**

<b>Evaluation Criteria Group</b>	<b>Group Ranking and Weight</b>
Reliable Design	8.52
Safety and Public Acceptance	7.71
Acceptable Program Risk	7.60
Attractive Costs	6.95
Compatible with Satellite	6.61
Flexible/Scalable Design	6.20
Attractive Lifetime	5.98
Flexible Operation	5.59
Enhance Survivability	5.32
Easy to Build, Handle, and Maintain	5.29
Schedule	5.25

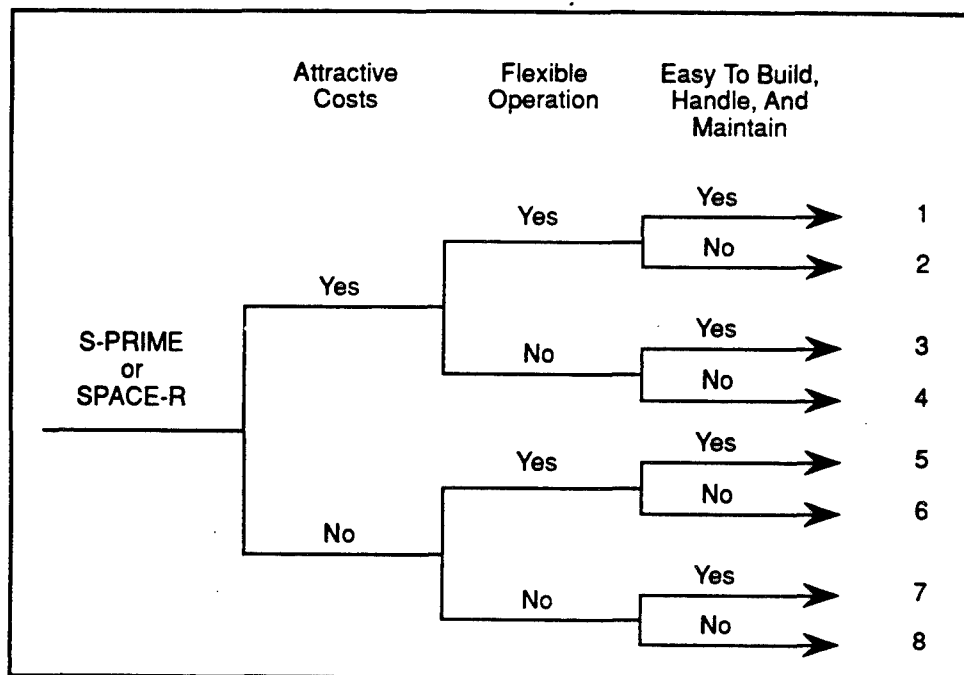
### **3.2 ISPA Event Tree Model**

The ISPA model for the evaluation of space nuclear reactor design concepts was developed in three stages. The first stage was to develop an event tree with the eleven evaluation criteria (referred to as Level 1 evaluation criteria) listed in Table 1. Figure 1 shows a small part of the event tree containing the first three evaluation criteria.

Each sequence in the event tree has an associated frequency. The frequency indicates the probability that the outcome is characterized by the combination of successes and failures that characterize that sequence. Quantifying the tree (or evaluating each sequence frequency) is performed by multiplying each of the event node branch split fractions together. As described earlier, the branch split fractions indicate the analysts' confidence that the outcome indicated by the corresponding branch is indeed correct. If the split fractions for each of the three top events shown in Figure 1 were 0.5, the frequency of each sequence (labeled 1 through 8) would be  $0.5^3$  (0.125).

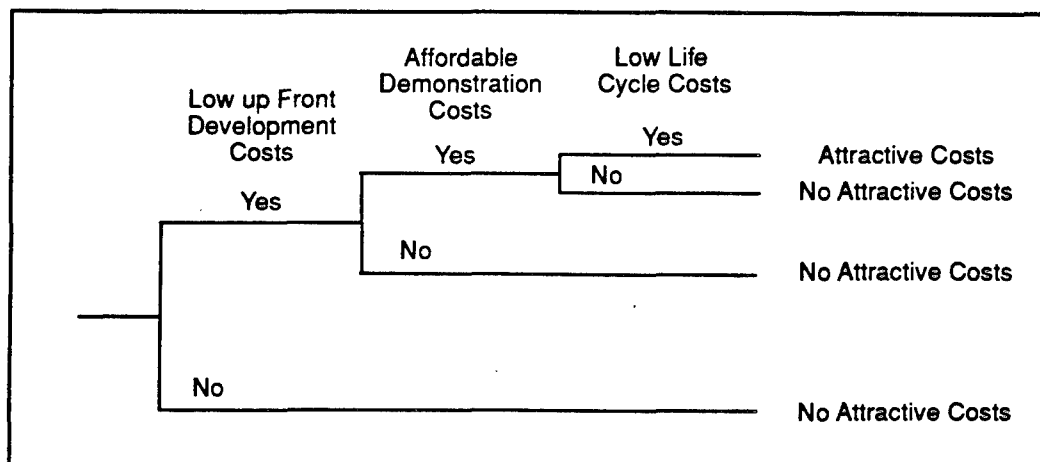
Stage 2 of the ISPA model development was to enhance the event tree by including sub-event trees to evaluate the top event (evaluation criteria) split fractions. To evaluate the success or failure of the evaluation criteria top events, the original "wants" used to establish the criteria were used.

Table 1 listed the evaluation criteria groups and the associated "wants" contained within each group. The logic structure input to the ISPA model permitted success of the group evaluation criteria only if each of the "wants" contained within that group were also successful.



**Figure 1. First Three Evaluation Criteria in Event Tree Format**

Figure 2 illustrates a subtree model for the first evaluation criteria (Attractive Costs).



**Figure 2. Attractive Costs Subtree**

By assigning or calculating a split fraction for each of the event nodes (referred to as Level 2 evaluation criteria) shown in Figure 2, the probability of successfully achieving Attractive Costs can be calculated. This success probability is then assigned to the "yes" branch of the Attractive Costs top event (shown in Figure 1).

The final stage of the ISPA model development was to include the physical attributes necessary to evaluate the success or failure of the Level 2 evaluation criteria. Twenty-eight design parameters were identified through a review of the two proposed design concepts. A list of dependencies was then generated to correlate the physical attributes with the Level 1 and Level 2 evaluation criteria. Table 3 lists the evaluation criteria (Level 1 and Level 2) and provides a cross-reference to the dependent attributes. Table 3 also provides a cross-reference of dependencies between the evaluation criteria. An example of this type of dependency is the requirement of achieving both Attractive Costs and Schedule to achieve an Acceptable Program Risk.

The ISPA model developed through the three-stage process just described is presented in Appendix A of this document. The EVNTRE input deck and the associated binners are listed in Appendix B. The model is a completely integrated performance assessment tool that uses the results of the QFD process as weighted evaluation criteria and systematically considers the integrated performance of nearly thirty components, systems, or design attributes. Although the model is fully integrated, the results are presented in a modular, easy-to-read format (the graphical representation presented in Appendix A). Figure 3 illustrates the general framework of the ISPA model.

### 3.3 Quantifying Performance

The final step in quantifying overall system performance is to score each event tree sequence based on the success or failure of the evaluation criteria. The magnitude of the score of an individual sequence is an indicator of the degree to which that sequence represents a favorable result (i.e., the higher the score, the more desirable the result). The Phase I ISPA model for the evaluation of space nuclear reactor design concepts includes eleven evaluation criteria (listed in Table 2 with a measure of relative importance). The analyst conducting the performance assessment has virtually unlimited flexibility regarding the scoring of sequences. For this assessment, a simple scoring scheme was used that scored individual branches in a sequence with their appropriate weighting factor (from Table 2) if the branch indicated success. If the branch indicated a failure with respect to a particular evaluation criteria, a score of zero was assumed. The total score for each sequence is evaluated by summing the scores of the individual branches under the evaluation criteria top events in each

sequence. For example, the sequence represented by a success for each evaluation criteria will have the maximum score of 71.02 (the total of the evaluation criteria weighting factors).

The overall performance of the system is evaluated by multiplying each sequence score by its associated frequency and summing over all sequences. Figure 4 shows a schematic representation of an ISPA model and how overall system performance is quantified.

Table 3. Cross-Reference of Design Attributes and Level 1 and Level 2 Evaluation Criteria

Attributes	Attractive Costs			Flexible Operation				Easy to Build, Handle, and Maintain			
	Low Up Front Development Costs	Affordable Demonstration Costs	Low Life Cycle Costs	Start-Up and Forget	Off Design Power Operation	Multiple Startups and Shutdowns	Operate in a Variety of Environments	Easy to Fabricate	Transportable	Repairable Up Until Launch	Repairable in Orbit
Proven Technology (TFEs)	/	/						/			
Proven Technology (Core Components)	/	/						/			
Redundancy			/	/		/					
TFE Costs		/	/								
Telemetry											
Control Sensors and Controller											
Ground Control				/							
Small Signature							/				
Specific Radiator Size							/		/		
Modularity								/	/	/	/
Fission Product Release										/	/
Shield Performance											/
Monitoring and Control											
Specific Power											
Standard Connections											
Structural Integrity									/		
Armor											
Reliability of Power Subsystem											
Reliability of Primary Coolant System											
Reliability of Reactivity Control System											
Reliability of Heat Removal System											
Reliability of Shield/Structural System											
Reliability of Supporting Systems											
Neutron Spectrum (fast of thermal)											



Table 3. Cross-Reference of Design Attributes and Level 1 and Level 2 Evaluation Criteria (continued)

	Attractive Costs			Flexible Operation				Easy to Build, Handle, and Maintain			
	Low Up Front Development Costs	Affordable Demonstration Costs	Low Life Cycle Costs	Start-Up and Forget	Off Design Power Operation	Multiple Startups and Shutdowns	Operate in a Variety of Environments	Easy to Fabricate	Transportable	Repairable Up Until Launch	Repairable in Orbit
Attributes											
TFE Design (multi- or single cell)											
Shutdown Contingencies (Physics Calculations)											
Safety Requirements											
Level 2 Evaluation Criteria											
Multiple Startups and Shutdowns					✓						
Demonstrable Reliability			✓		✓	✓					
Scalable Power											
Radiological Risk											
Key Technology Demonstrated by 10/94											
Development Risk											
Level 1 Evaluation Criteria											
Attractive Cost											
Flexible Operation											
Flexible/Scalable Design											
Safety and Public Acceptance											
Schedule											

Table 3. Cross-Reference of Design Attributes and Level 1 and Level 2 Evaluation Criteria (continued)

	Compatible With Satellite/Mission Objectives								Enhance Survivability of Satellite				Reliable Design	
	Minimal Negative Impact on Satellite	Improve Operational Performance of Satellite	Allow Use of Smaller Launch Vehicle	Ability to Support Secondary Function	Standard Interface for Coupling to Satellite	Avoid Detection	Avoid Attack	Survive Attack	Operate Through Attack	Demonstrable Reliability	No Credible Mission-Ending Single Point Failure			
Attribute														
Proven Technology (TFEs)														
Proven Technology (Core Components)														
Redundancy								✓	✓		✓			
TFE Costs														
Telemetry														
Control Sensors and Controller														
Ground Control														
Small Signature							✓							
Specific Radiator Size	✓		✓			✓								
Modularity														
Fission Product Release	✓			✓										
Shield Performance	✓													
Monitoring and Control		✓												
Specific Power			✓											
Standard Connections				✓	✓									
Structural Integrity							✓	✓						
Armor								✓	✓					
Reliability of Power Subsystem										✓				
Reliability of Primary Coolant System										✓				
Reliability of Reactivity Control System										✓				
Reliability of Heat Removal System										✓				
Reliability of Shield/Structural System										✓				
Reliability of Supporting Systems										✓				

Table 3. Cross-Reference of Design Attributes and Level 1 and Level 2 Evaluation Criteria (continued)

	Compatible With Satellite/Mission Objectives						Enhance Survivability of Satellite				Reliable Design	
	Minimal Negative Impact on Satellite	Improve Operational Performance of Satellite	Allow Use of Smaller Launch Vehicle	Ability to Support Secondary Function	Standard Interface for Coupling to Satellite	Avoid Detection	Avoid Attack	Survive Attack	Operate Through Attack	Demonstrable Reliability	No Credible Mission-Ending Single Point Failures	
Attributes												
Neutron Spectrum (fast of thermal)												
TFE Design (multi- or single cell)												
Shutdown Contingencies (Physics Calcs)												
Safety Requirements												
Level 2 Evaluation Criteria												
Multiple Startups and Shutdowns	✓			✓								
Demonstrable Reliability												
Scalable Power	✓			✓								
Radiological Risk												
Key Technology Demonstrated by 10/94												
Development Risk												
Level 1 Evaluation Criteria												
Attractive Cost												
Flexible Operation												
Flexible/Scalable Design												
Safety and Public Acceptance												
Schedule												

Table 3. Cross-Reference of Design Attributes and Level 1 and Level 2 Evaluation Criteria (continued)

	Flexible/Scalable Design								Safety and Public Acceptance			Schedule	
	Scalable Over Small Power Range With No Design Changes	Scalable Over Medium Power Range With No Technology or Component Design Changes	Scalable Technology Over Large Power Range	Simplicity - Use Common Components Within Same Reactor	Simplicity - Use Common Components While Scaling Power	Ability to Easily Incorporate Technological Advancements	Low Radiological Risk to Biosphere	Sealable to Public (Safety, Cost, Benefit)	Man-Rated	Key Technologies Demonstrated by 10/94	Fly by 10/98		
Attributes													
Proven Technology (TFEs)							✓			✓			
Proven Technology (Core Components)										✓			
Redundancy													
TFE Costs													
Telemetry													
Control Sensors and Controller													
Ground Control													
Small Signature													
Specific Radiator Size													
Modularity				✓	✓	✓							
Fission Product Release							✓						
Shield Performance													
Monitoring and Control													
Specific Power													
Standard Connections													
Structural Integrity							✓						
Armor													
Reliability of Power Subsystem													
Reliability of Primary Coolant System													
Reliability of Reactivity Control System													
Reliability of Heat Removal System													
Reliability of Shield/Structural System													
Reliability of Supporting Systems													

Table 3. Cross-Reference of Design Attributes and Level 1 and Level 2 Evaluation Criteria (continued)

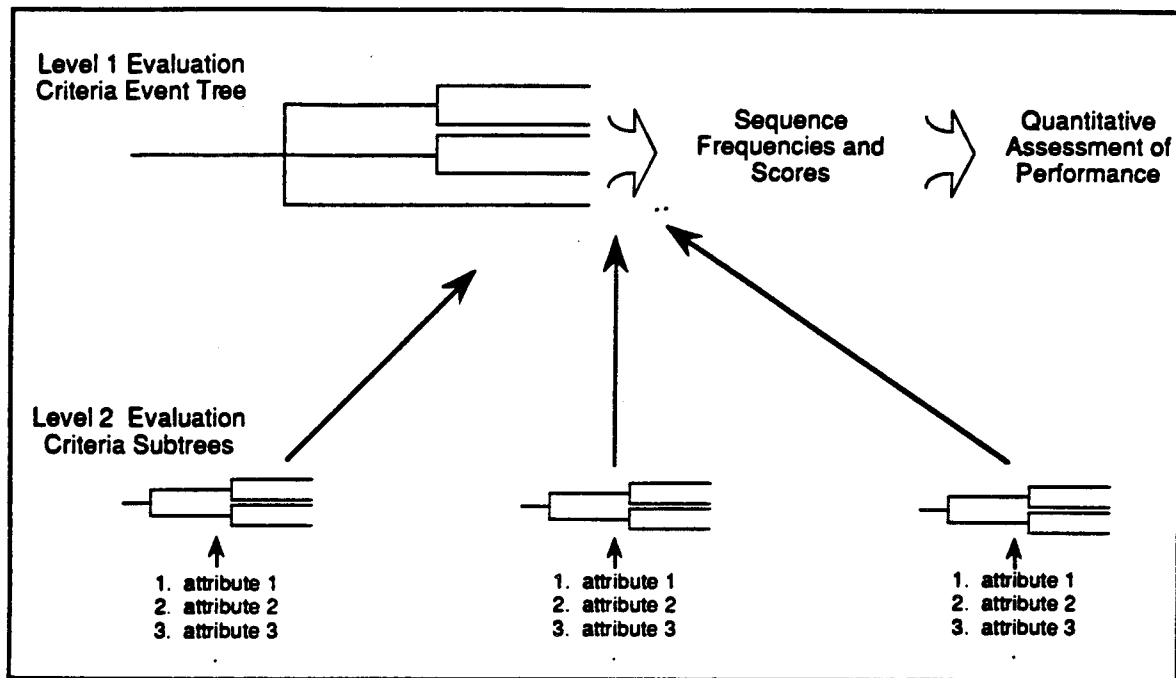
	Flexible/Scalable Design							Safety and Public Acceptance			Schedule	
	Scalable Over Small Power Range With No Design Changes	Scalable Over Medium Power Range With No Technology or Component Design Changes	Scalable Technology Over Large Power Range	Simplicity – Use Common Components Within Same Reactor	Simplicity – Use Common Components While Scaling Power	Ability to Easily Incorporate Technological Advancements	Low Radiological Risk to Biosphere	Sealable to Public (Safety, Cost, Benefit)	Man-Rated	Key Technologies Demonstrated by 10/84	Fly by 10/88	
<b>Attributes</b>												
Neutron Spectrum (fast of thermal)	✓	✓	✓									
TFE Design (multi- or single cell)	✓	✓	✓		✓							
Shutdown Contingencies (Physics Calcs)							✓					
Safety Requirements							✓					
<b>Level 2 Evaluation Criteria</b>												
Multiple Startups and Shutdowns												
Demonstrable Reliability												
Scalable Power												
Radiological Risk								✓			✓	
Key Technology Demonstrated by 10/84												
Development Risk												
<b>Level 1 Evaluation Criteria</b>												
Attractive Cost								✓			✓	
Flexible Operation												
Flexible/Scalable Design												
Safety and Public Acceptance											✓	
Schedule												

Table 3. Cross-Reference of Design Attributes and Level 1 and Level 2 Evaluation Criteria (continued)

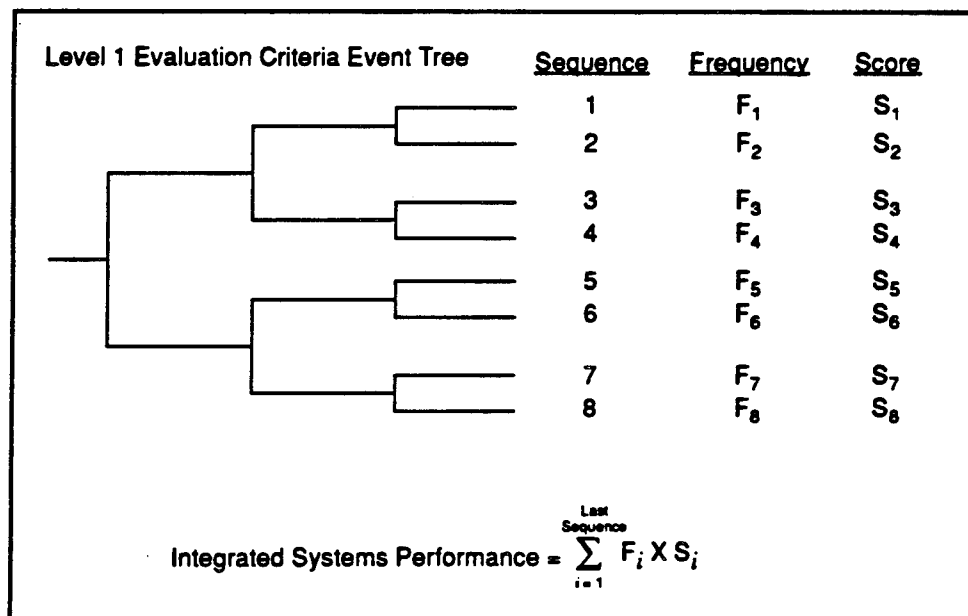
	Acceptable Program Risk					Attractive Lifetime	
	Acceptable Development Risk High Probability That Development Will Be Successful – Technical, Cost, Schedule)	Performance Must Be Assessable While Satellite is in Orbit	Testable – Full System Qualification and Acceptance	Long Operational Life	Lifetime Tailorable to Mission		
Attributes							
Proven Technology (TFEs)	✓						
Proven Technology (Core Components)	✓						
Redundancy							
TFE Costs							
Telemetry		✓					
Control Sensors and Controller		✓					
Ground Control		✓					
Small Signature							
Specific Radiator Size							
Modularity							
Fission Product Release							
Shield Performance							
Monitoring and Control							
Specific Power							
Standard Connections							
Structural Integrity							
Armor							
Reliability of Power Subsystem							
Reliability of Primary Coolant System							
Reliability of Reactivity Control System							
Reliability of Heat Removal System							
Reliability of Shield/Structural System							
Reliability of Supporting Systems							

Table 3. Cross-Reference of Design Attributes and Level 1 and Level 2 Evaluation Criteria (continued)

Acceptable Program Risk					Attractive Lifetime	
	Acceptable Development Risk (High Probability That Development Will Be Successful - Technical, Cost, Schedule)	Performance Must Be Assessable While Satellite is in Orbit	Testable - Full System Qualification and Acceptance	Long Operational Life	Lifetime Tailorable to Mission	
Attributes						
Neutron Spectrum (fast of thermal)						
TFE Design (multi- or single cell)			✓			
Shutdown Contingencies (Physics Cells)						
Safety Requirements						
Level 2 Evaluation Criteria						
Multiple Startups and Shutdowns						
Demonstrable Reliability	✓					
Scalable Power						
Radiological Risk						
Key Technology Demonstrated by 10/94						
Development Risk						
Level 1 Evaluation Criteria						
Attractive Cost	✓					
Flexible Operation				✓	✓	
Flexible/Scalable Design					✓	
Safety and Public Acceptance						
Schedule	✓					



**Figure 3. General Framework of the ISPA Model**



**Figure 4. Quantifying Overall System Performance**



## 4.0 RESULTS OF PERFORMANCE ASSESSMENT

Prior to our discussion of the performance assessment results, it is important to note that the split fractions assigned to the branch points representing the physical attributes of this Phase I ISPA model are somewhat arbitrary. The various probabilities that the two space nuclear reactor design concepts will achieve individual design goals have been input by SAIC based on only a limited review of the proposed concepts. Our purpose in this exercise was to illustrate the ISPA concept and produce a model that could provide the starting point for more rigorous evaluation. Therefore, the use of these results as an indicator of performance (i.e., comparing the results of the S-PRIME assessment to the SPACE-R assessment) is inappropriate pending a detailed evaluation of both concepts and formal quantification of the attribute split fractions (Phase II of the ISPA project). However, sensitivity/uncertainty assessments are possible and are provided below to demonstrate the potential of the model to provide insights regarding the importance of the various systems on overall performance.

### 4.1 Importance Assessment

Table 4 lists the results of the ISPA evaluation for the two proposed space nuclear reactor design concepts; Table 5 lists the success frequencies for each of the eleven evaluation criteria. The data presented in Table 5 provide valuable insights regarding the importance of the evaluation criteria and their potential impact on the overall system performance. The evaluation criterion with the highest weight (or importance) is *Reliable Design*. For both the S-PRIME and SPACE-R concepts, the success frequency is quite high<sup>2</sup> (>90 percent). Because the success frequency is so high, any additional increase in reliability will not likely result in a significant increase in overall system performance. However, looking at the evaluation criterion ranked second (e.g., *Safety and Public Acceptance*), we can see that the success frequency for both design concepts is quite low (~7 to 10 percent). Unlike the reliability evaluation criteria, any increase in the success frequency of the *Safety and Public Acceptance* top event will likely result in a significant increase in overall system performance. This simple application of the ISPA methodology allows the analyst to identify and

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2 A design reliability of 95 percent was required by the design specification. Each designer thus attributed this level of reliability to their design. No separate evaluation of this result was made under this effort. The deviation from the original figure results from the subjective assessments made under this effort.

rank the importance or components as they relate to the evaluation criteria, thus providing guidance for the allocation of additional resources or analyses in the most effective manner.

**Table 4. Base Case ISPA Results for S-PRIME and SPACE-R**

Design	ISPA Result
S-PRIME	16.63
SPACE-R	17.95

**Table 5. Success Frequency for the Eleven Evaluation Criteria**

	SPACE-R	S-PRIME
Evaluation Criteria	Success Frequency	Success Frequency
Attractive Costs	0.534	0.341
Flexible Operation	0.189	0.171
Easy to Build, Handle, and Maintain	0.158	0.159
Flexible/Scalable Design	0.208	0.354
Compatible with Satellite/Mission Objectives	0.04	0.07
Enhance Survivability of Satellite	0.12	0.108
Reliable Design	0.92	0.912
Safety and Public Acceptance	0.108	0.069
Schedule	0.11	0.069
Acceptable Program Risk	0.09	0.028
Attractive Lifetime	0.04	0.063

## 4.2 Sensitivity Assessment

Four additional assessments were performed to determine the sensitivity of the overall system performance to uncertain parameters. The SPACE-R concept was arbitrarily selected for the

sensitivity assessment; the results are summarized in Table 6. The first sensitivity assessment involved decreasing confidence from 0.95 to 0.5 that the TFE technology is proven. The result was a decrease in overall system performance of 17.3 percent. This significant decrease in the system's performance occurs because of the importance (i.e., high weight assigned) of the *Proven Technology (TFEs)* attribute among the evaluation criteria. Referring to Table 3, we can see that the success or failure of six Level 2 evaluation criteria is dependent on proven TFE technology. These six Level 2 evaluation criteria then affect the success or failure of the following Level 1 evaluation criteria:

- Attractive Costs.
- Easy to Build, Handle, and Maintain.
- Safety and Public Acceptance.
- Schedule.
- Acceptable Program Risk.

**Table 6. Results of SPACE-R Sensitivity Assessment**

Performance Assessment	Score	Change Relative to Base Case
SPACE-R Base Case	17.95	--
<i>Sensitivity Cases</i>		
Decrease confidence that the TFE technology is proven (from 0.95 to 0.5)	14.84	-17.3%
Decrease confidence that the power subsystem will be reliable (from 0.982 to 0.8)	15.11	-15.8%
Increase confidence that the reactor will have a small signature (from 0.5 to 1.0)	19.88	+ 10.8%
Increase confidence that there will be mission-ending single point failures (from 0.01 to 0.1)	17.2	-4%
Increase confidence from 0.5 to 0.75 that TFE costs are low and simultaneously decrease confidence that there are no mission-ending single point failures from 0.01 to 0.1	18.39	+ 2.5%

The second sensitivity assessment once again demonstrates the importance that a single attribute can have on overall system performance. For this calculation, the confidence in the reliability of the power subsystem was decreased from 0.982 to 0.8. This resulted in a decrease in system performance of 15.8 percent.

The final sensitivity assessment was designed to reflect a somewhat more realistic situation in which the effect of multiple dependencies on the overall system performance is not intuitively obvious. The assessment assumed that a proposed decrease in TFE costs (an increase in confidence from 0.5 to 0.75 that the TFE costs will be low) would result in an order of magnitude increase in the probability of a mission-ending single point failure (assumed to be some common mode failure of the TFEs). Because of the multiple levels of dependencies (TFE costs affect the Attractive Costs evaluation criteria that in turn affects several other evaluation criteria, such as Safety and Public Acceptance, and Acceptable Program Risk) and the different weights applied to the various evaluation criteria, the effect of this design trade-off on overall system performance would be extremely difficult (if not impossible) to assess without an integrated model. The result of the ISPA evaluation showed a 2.5 percent increase in overall system performance.

## 5.0 SUMMARY OF PHASE I ISPA

The goal of Phase I of the PL ISPA project to evaluate space nuclear reactor design concepts is to demonstrate the ISPA methodology and the ability to perform integrated performance assessments where multiple dependencies and significant uncertainties exist. An ISPA model has been developed to evaluate two proposed space nuclear reactor design concepts: S-PRIME and SPACE-R. The weighted criteria for evaluating the performance of the two conceptual designs are based on the results of a QFD exercise in which expert elicitations were used to identify and rank "wants" in a thermionic space power system. The ISPA model considers twenty-eight design attributes and evaluates the success or failure of the evaluation criteria based on the level of confidence the analysts place in each concept to achieve specific design goals. The model is capable of quantitatively evaluating the performance of each design concept in a systematic, self-consistent manner. Dependencies both between the criteria and between components included in the design are represented in the model.

At this phase of the project, a thorough review and independent assessment of the two proposed design concepts has not been performed. The confidence level split fractions input to the model (e.g., the likelihood that the heat rejection system will be reliable, the likelihood that the system will be fully testable, etc.) are estimates based on a preliminary review of the proposed designs. Pending a thorough design review and formal quantification of the ISPA model, any comparison of the S-PRIME performance to the SPACE-R performance based on the material presented herein is inappropriate. However, relative changes in overall performance can be assessed for the purposes of demonstrating the methodology and identifying and evaluating the importance of sensitive or uncertain parameters.

Initial results of the ISPA evaluation indicate that significant impacts on overall system performance can result from relatively small perturbations in confidence related to evaluation criteria that have a low probability of success. Conversely, only limited performance effects will result from relatively large perturbations in confidence related to criteria that initially have a high probability of success. These results provide the analyst with insights into the most effective way to allocate time or resources in efforts associated with improving the design of a SNPS.

The most effective demonstration of the ISPA methodology is in its application as a decision analysis tool to a realistic performance assessment problem. Calculations have been performed that demonstrate the model's ability to effectively quantify the net performance increase or decrease from

proposed design changes where those changes influence multiple systems (either in a positive or negative manner) that are not of equal importance. Quantifying the effect of complex design changes or performing design trade-off studies would be extremely difficult without an integrated model.

## 6.0 PHASES II AND III ISPA

As stated in Section 4.0 of this report, the attribute split fractions (indicating the level of confidence that the outcome is indeed correct) input to the Phase I ISPA model are based on a preliminary review of the two proposed space nuclear reactor design concepts. The practical value of the ISPA approach will only be realized when results obtained from the mechanistic modeling tools applied by the PL (e.g., heat, pipe models, structural analysis, and core neutronics evaluation) are used within this framework. This merger will facilitate a realistic evaluation of the conceptual designs. Use of an integrated framework in this evaluation will help ensure a systematic and self-consistent evaluation of each design. The fact that both designs are evaluated within the same framework will help ensure that the potential for analytical bias will be reduced. Finally, the ability to reflect both the PL's evaluation and the designer's evaluation within a single model framework will allow direct assessment of the impact of modeling differences. This will assist in the direction of future research and development resources toward areas that will provide the greatest payback in terms of reducing technology risk.

Phase II of the ISPA project involves using the model developed under Phase I to quantitatively evaluate the two concepts. This task will involve the following: (1) a thorough review of the ISPA model by PL personnel to ensure that all relevant design attributes and evaluation criteria have been included; (2) a review of the design attribute/evaluation criteria cross-reference matrix to ensure that all dependencies are being modeled correctly; and (3) an analytical evaluation of the two proposed design concepts to facilitate quantification of the design attribute split fractions. Once these three tasks are completed, final quantitative ISPA evaluations will be made. These results will be suitable for direct comparison and should provide PL with valuable insights regarding final selection and design enhancements of a space nuclear reactor concept.

Phase III of the ISPA project involves developing an enhanced ISPA model for one (or both) of the concepts being evaluated. This will be performed by refining the Phase II ISPA model to more accurately represent systems or phenomena identified as sensitive regarding the final performance evaluation. The Phase III enhanced model will be used for complete evaluation of a selected design and the evaluation of design trade-offs. In addition to evaluation of the reactor design concepts, the Phase III ISPA model will also be used to support assessments of the following:

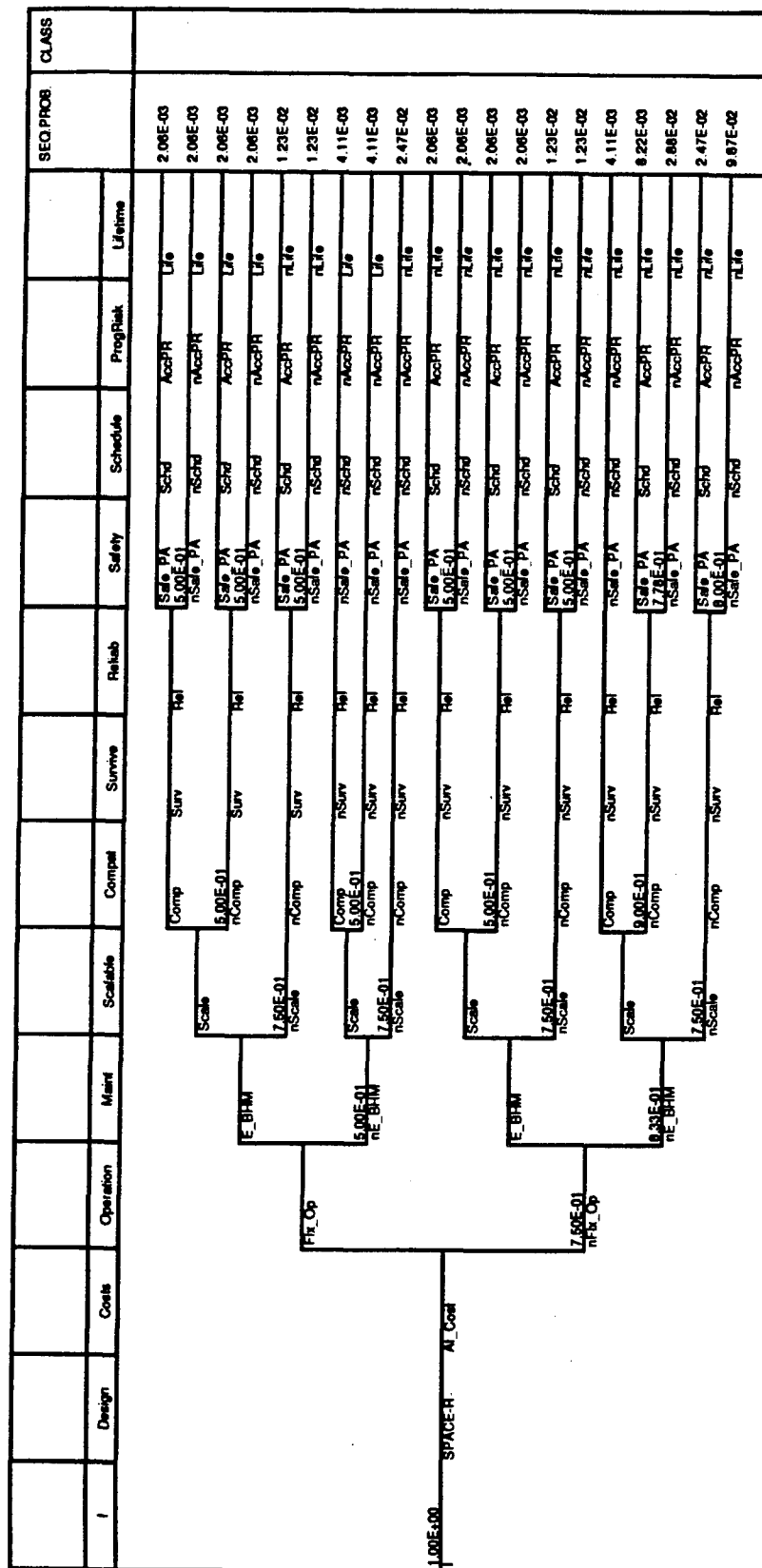
- Design innovations to extend useful lifetime.
- Design changes to enhance reliability.

- Performance with respect to a range of mission applications.
- Effects of design choices on development costs.
- Testing risk, cost, and schedule.

In summary, the Phase I ISPA model should be regarded only as a proof-of-concept study. In keeping with this objective, costs have been held to a minimum. However, the real payback to this modeling initiative will be realized in the actual application.



# APPENDIX A - ISPA MODEL



SNPS - ISPA EVALUATION MODEL SNPS01.TRE 10-14-92

Figure A-1. Level 1 Evaluation Criteria Event Tree





I	Design	TFE_Ret	Cor_Ret	Rad_Size	Modular	Fabric	Strc_In	Transp	Rep_Low	FP_Ret	Shield	Rep_Obs	Blm	SEQ PROB	CLASS
1.00E+00	SPACER	P_TFE	P_Core	Sm_Had	Mod	E_Fab	ST_Ing	Trans	Rep_Low	FP_Ret	Shield	Rep_Obs	E_BRM	1.59E-01	
				nsim_Had	nsimod	ns_Fab	ns_Ing	ns	ns	ns	ns	ns	ns	1.28E-03	
				nsim_Had	nsimod	ns_Fab	ns_Ing	ns	ns	ns	ns	ns	ns	1.67E-02	
				nsim_Had	nsimod	ns_Fab	ns_Ing	ns	ns	ns	ns	ns	ns	1.79E-01	
				nsim_Had	nsimod	ns_Fab	ns_Ing	ns	ns	ns	ns	ns	ns	1.73E-02	
				nsim_Had	nsimod	ns_Fab	ns_Ing	ns	ns	ns	ns	ns	ns	3.52E-01	
				nsim_Had	nsimod	ns_Fab	ns_Ing	ns	ns	ns	ns	ns	ns	1.73E-02	
				nsim_Had	nsimod	ns_Fab	ns_Ing	ns	ns	ns	ns	ns	ns	1.79E-01	
				nsim_Had	nsimod	ns_Fab	ns_Ing	ns	ns	ns	ns	ns	ns	4.37E-02	

SNPS - ISPA EVALUATION MODEL BHM01.TRE 10-14-92

Figure A-4. Easy to Build, Handle, and Maintain Subtree



I	Design	Rad_Src	FP_Rad	Shield	Min_Impcd	MaxCon	Max_SuSd	Scale	Imp_OpPr	SpdPwr	LaunchV	Sat_Con	Sec_Func	Sat_Int	Comp	SEQ PROB	CLASS
1.00E+00	SPACE-H	Sm_Had 6.00E-01 nSm_Had	1.77E+02 h_TPRad	7.00E-03 nShield	MAN_TMP nMAN_TMP	MANCon nMANCon	MAN_SU nMAN_SU	Scale 7.00E-01 nScale	Imp_Pwr nImp_Pwr	9.71E-02 h_SpPr	Sm_LV nSm_LV	Sat_Con nSat_Con	Sec_Fun nSec_Fun	Sat_Int nSat_Int	Comp nComp	4.00E-02 4.30E-03 1.00E-01 9.31E-03 2.17E-01 3.00E-03 4.40E-02 4.81E-01	

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Figure A-6. Compatible with Satellite Subtree



				SEQ.PROB.	CLASS
I	Dem_Rel	ME_SPFs	Reliab		
1.00E+00	Dem_Rel	nME_SPFs	Rel	9.22E-01	
I	3.27E-02	ME_SPFs	nRel	6.48E-03	
	nDem_Rel	Irr_SPFs	nRel	3.27E-02	

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Figure A-8. Reliable Design Subtree











# APPENDIX B - EVNTRE INPUT DECK

## SNPS - ISPA EVALUATION MODEL

77

NQ

1 1.000

### SNPS Design Assessment

#### 1 What is Design Concept?

2 S-PRIME SPACE-R

1 1 2

0.000 1.000

#### 2 Is the TFE design multi- or single-cell?

2 S\_TFE M\_TFE

2 1 2

3

1 1

1

S-PRIME

0.000 1.000

1 1

2

SPACE-R

1.000 0.000

Otherwise - Not Used

1.000 0.000

#### 3 Is the neutron spectrum fast, thermal, or either?

3 fSpec tSpec eSpec

2 1 2 3

3

1 1

1

S-PRIME

0.000 0.000 1.000

1 1

2

SPACE-R

0.000 0.000 1.000

Otherwise - Not Used

1.000 0.000 0.000

#### 4 Is the TFE technology proven?

2 P\_TFE nP\_TFE

2 1 2

3

1	1		
	1		
	S-PRIME		
	0.900	0.100	
1	1		
	2		
	SPACE-R		
	0.950	0.050	
	Otherwise - Not Used		
	1.000	0.000	

5 Is the core component technology proven?

2	P_Core	nP_Core
2	1	2
3		
1	1	
	1	
	S-PRIME	
	0.900	0.100
1	1	
	2	
	SPACE-R	
	0.800	0.200
	Otherwise - Not Used	
	1.000	0.000

6 What is the level of redundancy?

2	h_redun	l_redun
2	1	2
3		
1	1	
	1	
	S-PRIME	
	0.900	0.100
1	1	
	2	
	SPACE-R	
	0.990	0.010
	Otherwise - Not Used	
	0.000	1.000

7 What is the reliability of the power subsystem?

2	Rel_Pwr	nRel_Pwr
2	1	2
3		

	1	1	
		1	
		S-PRIME	
		0.997	0.003
	1	1	
		2	
		SPACE-R	
		0.982	0.018
		Otherwise - Not Used	
		1.000	0.000
8 What is the reliability of the primary coolant system?			
	2	Rel_NaK	nRel_NaK
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.995	0.005
\$Case	1		
	1	1	
		2	
		SPACE-R	
		0.997	0.003
\$Case	2		
		Otherwise - Not Used	
		1.000	0.000
\$Case	3		
9 What is the reliability of the reactivity control system?			
	2	Rel_RC	nRel_RC
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.963	0.037
\$Case	1		
	1	1	
		2	
		SPACE-R	
		0.983	0.017
\$Case	2		

Otherwise - Not Used

1.000 0.000

\$Case 3

10 What is the reliability of the heat removal components?

2 Rel\_HX nRel\_HX

2 1 2

3

1 1

1

S-PRIME

0.995 0.005

\$Case 1

1 1

2

SPACE-R

0.997 0.003

\$Case 2

Otherwise - Not Used

1.000 0.000

\$Case 3

11 What is the reliability of the shield and structures?

2 Rel\_SS nRel\_SS

2 1 2

3

1 1

1

S-PRIME

1.000 0.000

\$Case 1

1 1

2

SPACE-R

0.997 0.003

\$Case 2

Otherwise - Not Used

1.000 0.000

\$Case 3

12 What is the reliability of the supporting systems  
(wiring, TFE inte

2 Rel\_Sup nRel\_Sup

2 1 2

3



	1	1				
		1				
		S-PRIME				
		0.999	0.001			
\$Case	1					
	1	1				
		2				
		SPACE-R				
		0.995	0.005			
\$Case	2					
		Otherwise - Not Used				
		1.000	0.000			
\$Case	3					
	13	What is the demonstrable reliability of the design?				
	2	Dem_Rel	nDem_Rel			
	2	1	2			
	2					
	6	7	8	9	10	11
12						
		1	1	1	1	1
1						
		Rel_Pwr	Rel_NaK	Rel_RC	Rel_HX	Rel_SS
Rel_Sup						
		1.000	0.000			
\$Case	1					
		Otherwise - No Demonstrable Reliability				
		0.000	1.000			
\$Case	2					
	14	What are the TFE costs?				
	2	l_Tcost	h_Tcost			
	2	1	2			
	3					
	1	2				
		1				
		S_TFE				
		0.750	0.250			
\$Case	1					
	1	2				
		2				
		M_TFE				
		0.500	0.500			
\$Case	2					

Otherwise - Not Used

0.000 1.000

\$Case 3

15 Are the up front development costs low?

2 l\_dvcost h\_dvcost

2 1 2

2

2 4 5

1 1

P\_TFE P\_Core

1.000 0.000

\$Case 1

Otherwise - High Up Front Development Costs

0.000 1.000

\$Case 3

16 Are the demonstration costs affordable?

2 l\_dmcost h\_dmcost

2 1 2

2

2 4 5

1 1

P\_TFE P\_Core

1.000 0.000

\$Case 1

Otherwise - High Demonstration Costs

0.000 1.000

\$Case 3

17 Are the life cycle costs < 500/We

2 l\_lcost h\_lcost

2 1 2

2

3 6 13 14

1 1 1

h\_redun Dem\_Rel l\_Tcost

1.000 0.000

\$Case 1

Otherwise - High Life Cycle Costs

0.000 1.000

\$Case 2

18 How attractive are the costs?

2 At\_Cost nAt\_Cost

2 1 2

	2			
	3	15	16	17
		1	1	1
		1_dvcost	1_dmcost	1_lcost
		1.000	0.000	

\$Case 1

Otherwise - Costs Not Attractive

0.000 1.000

\$Case 2

19 Is there adequate telemetry?

	2	Tele	nTele
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.900	0.100

\$Case 1

	1	1
		2
		SPACE-R
		0.900 0.100

\$Case 2

Otherwise - Not Used

0.000 1.000

\$Case 3

20 Are the control sensors and controller reliable?

	2	Cont	nCont
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.900	0.100

\$Case 1

	1	1
		2
		SPACE-R
		0.900 0.100

\$Case 2

Otherwise - Not Used

		0.000	1.000
\$Case	3		
	21	Are there means for adequate ground control?	
	2	GrndCon	nGrndCon
	2	1	2
	2		
	2	19	20
		1	1
		Telem	Cont
		1.000	0.000
\$Case	1		
		Otherwise - Not Adequate Ground Control	
		0.000	1.000
\$Case	2		
	22	Does the reactor have a small signature?	
	2	Sm_Sig	nSm_Sig
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.500	0.500
\$Case	1		
	1	1	
		2	
		SPACE-R	
		0.500	0.500
\$Case	2		
		Otherwise - Not Used	
		0.000	1.000
\$Case	3		
	23	Is the specific radiator size small?	
	2	Sm_Rad	nSm_Rad
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.500	0.500
\$Case	1		
	1	1	
		2	

		SPACE-R	
		0.500	0.500
\$Case	2		
		Otherwise - Not Used	
		0.000	1.000
\$Case	3		
	24	Can the reactor operate in start-up and forget mode?	
	2	SU_FG	nSU_FG
	2	1	2
	2		
	2	6	21
		1	1
		h_redun	GrndCon
		1.000	0.000
\$Case	1		
		Otherwise - No Start-Up and Forget	
		0.000	1.000
\$Case	2		
	25	Is the reactor capable of multiple startups and shutdowns?	
	2	MSU_SD	nMSU_SD
	2	1	2
	2		
	2	6	13
		1	1
		h_redun	Dem_Rel
		1.000	0.000
\$Case	1		
		Otherwise - No Multiple Startups and Shutdowns	
		0.000	1.000
\$Case	2		
	26	Is the reactor capable of off-design power operation?	
	2	OD_Pwr	nOD_Pwr
	2	1	2
	2		
	2	13	25
		1	1
		Dem_Rel	MSU_SD
		1.000	0.000
\$Case	1		
		Otherwise - No Off-Design Power Operation	

	0.000	1.000		
\$Case 2				
27 Is the reactor capable of operating in a variety of environments?				
2 Var_Env nVar_Env				
2 1 2				
2				
2 22 23				
1 1				
Sm_Sig Sm_Rad				
1.000 0.000				
\$Case 1				
Otherwise - No Operation in a Variety of Environments				
0.000 1.000				
\$Case 2				
28 How flexible is the operation of the reactor?				
2 Flx_Op nFlx_Op				
2 1 2				
2				
4 24 25 26 27				
1 1 1 1				
SU_FG MSU_SD OD_Pwr Var_Env				
1.000 0.000				
\$Case 1				
Otherwise - No Flexible Operation				
0.000 1.000				
\$Case 2				
29 Is the design modular?				
2 Mod nMod				
2 1 2				
3				
1 1				
1				
S-PRIME				
0.900 0.100				
\$Case 1				
1 1				
2				
SPACE-R				
0.950 0.050				
\$Case 2				

Otherwise - Not Used

0.000 1.000

\$Case 3

30 Will the reactor be easy to fabricate?

2	E_Fab	nE_Fab	
2	1	2	
2			
3	4	5	29
	1	1	1
	P_TFE	P_Core	Mod
	1.000	0.000	

\$Case 1

Otherwise - Not Easy to Fabricate

0.000 1.000

\$Case 2

31 Does the reactor have structural integrity?

2	St_Intg	nSt_Intg	
2	1	2	
3			
1	1		
	1		
	S-PRIME		
	0.500	0.500	

\$Case 1

1	1	
	2	
	SPACE-R	
	0.500	0.500

\$Case 2

Otherwise - Not Used

0.000 1.000

\$Case 3

32 Is the reactor transportable?

2	Trans	nTrans	
2	1	2	
2			
3	23	29	31
	1	1	1
	Sm_Rad	Mod	St_Intg
	1.000	0.000	

\$Case 1

Otherwise - Not Transportable

		0.000	1.000
\$Case	2		
	33	Is the reactor repairable up until launch?	
	2	Rep_Lau	nRep_Lau
	2	1	2
	2		
	1	29	
		1	
		Mod	
		1.000	0.000
\$Case	1		
		Otherwise - Not Repairable Up Until Launch	
		0.000	1.000
\$Case	2		
	34	Will the fission product release be low?	
	2	l_FPRel	h_FPRel
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.900	0.100
\$Case	1		
	1	1	
		2	
		SPACE-R	
		0.900	0.100
\$Case	2		
		Otherwise - Not Used	
		0.000	1.000
\$Case	3		
	35	Will the shield performance be adequate?	
	2	Shield	nShield
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.990	0.010
\$Case	1		
	1	1	
		2	



		SPACE-R			
		0.990	0.010		
\$Case	2	Otherwise - Not Used			
		0.000	1.000		
\$Case	3	36 Is the reactor repairable in orbit?			
		2	Rep_Orb	nRep_Orb	
		2	1	2	
		2			
		3	29	34	35
			1	1	1
			Mod	1_FPRel	Shield
			1.000	0.000	
\$Case	1	Otherwise - Not Repairable In Orbit			
		0.000	1.000		
\$Case	2	37 Is the reactor easy to build, handle, and maintain?			
		2	E_BHM	nE_BHM	
		2	1	2	
		2			
		4	30	32	33 36
			1	1	1 1
			E_Fab	Trans	Rep_Lau Rep_Orb
			1.000	0.000	
\$Case	1	Otherwise - Not Easy to Build, Handle, and Maintain			
		0.000	1.000		
\$Case	2	38 Is the power scalable - small range?			
		2	Scal_s	nScal_s	
		2	1	2	
		2			
		2	3	3	
			(2	+ 3)	
			tSpec	eSpec	
			1.000	0.000	
\$Case	1	Otherwise - Not Scalable - Small Power Range			
		0.000	1.000		
\$Case	2				

```

39 Is the power scalable - medium range?
    2 Scal_m nScal_m
    2      1      2
    3
    2      3      2
      3      1
      eSpec S_TFE
      0.500 0.500
$Case 1
    2      3      2
      3      2
      eSpec M_TFE
      0.900 0.100
$Case 2
      Otherwise - Not Scalable - Medium Power Range
      0.000 1.000
$Case 3
40 Is the power scalable - large range?
    2 Scal_l nScal_l
    2      1      2
    2
    2      3      3
      (1 + 3)
      fSpec eSpec
      0.900 0.100
$Case 1
      Otherwise - Not Scalable - Large Power Range
      0.000 1.000
$Case 2
41 Simplicity - Use common components (CC) within same
reactor?
    2 SimpCC nSimpCC
    2      1      2
    2
    1      29
      1
      Mod
      1.000 0.000
$Case 1
      Otherwise - No Common Components Within Same Reactor
      0.000 1.000
$Case 3

```

42 Simplicity - Use common components while scaling power?

	2	SimpSP	nSimpSP	
	2	1	2	
	3			
	2	2	29	
		2	1	
		M_TFE	Mod	
		0.500	0.500	

\$Case 1

	2	2	29
		1	1
		S_TFE	Mod
		0.500	0.500

\$Case 2

Otherwise - No Common Components While Scaling Power  
0.000 1.000

\$Case 3

43 Can the design incorporate technological advances?

	2	Tech_A	nTech_A
	2	1	2
	2		
	1	29	
		1	
		Mod	
		1.000	0.000

\$Case 1

Otherwise - Can't Incorporate Technological Advances  
0.000 1.000

\$Case 2

44 Is the design flexible/scalable?

	2	Scale	nScale
	2	1	2
	2		
	6	38	39

43

40 41 42

1 1 1 1 1

1

Scal\_s Scal\_m Scal\_l SimpCC SimpSP

Tech\_A

1.000 0.000

\$Case 1

Otherwise - Not Flexible/Scalable

	0.000	1.000		
--	-------	-------	--	--

\$Case 2

45 Is there a minimal negative impact on the satellite?

2	MIN_IMP	nMIN_IMP		
2	1	2		
2				
3	23	34	35	
	1	1	1	
	Sm_Rad	1_FPRel	Shield	
	1.000	0.000		

\$Case 1

Otherwise - Negative Impact on Satellite

	0.000	1.000		
--	-------	-------	--	--

\$Case 2

46 Is there adequate monitoring and control?

2	MonCon	nMonCon		
2	1	2		
3				
1	1			
	1			
	S-PRIME			
	0.500	0.500		

\$Case 1

1	1			
	2			
	SPACE-R			
	0.500	0.500		

\$Case 2

Otherwise - Not Used

	0.000	1.000		
--	-------	-------	--	--

\$Case 3

47 Will the reactor improve the operational performance of the satelli

2	Imp_Per	nImp_Per		
2	1	2		
2				
3	25	46	44	
	1	1	1	
	MSU_SD	MonCon	Scale	
	1.000	0.000		

\$Case 1

Otherwise - No Improvement in Operational  
Performance

0.000 1.000

\$Case 2

48 Is the specific power high?

2 hSpPwr lSpPwr

2 1 2

3

1 1

1

S-PRIME

0.950 0.050

\$Case 1

1 1

2

SPACE-R

0.900 0.100

\$Case 2

Otherwise - Not Used

0.000 1.000

\$Case 3

49 Will the design allow use of a smaller launch vehicle?

2 Sm\_LV nSm\_LV

2 1 2

2

2 23 48

1 1

Sm\_Rad hSpPwr

1.000 0.000

\$Case 1

Otherwise - No Smaller Launch Vehicle

0.000 1.000

\$Case 2

50 Are the connections used standard?

2 Std\_Con nStd\_Con

2 1 2

3

1 1

1

S-PRIME

1.000 0.000

\$Case 1

	1	1			
		2			
		SPACE-R			
		1.000	0.000		
\$Case	2				
		Otherwise - Not Used			
		0.000	1.000		
\$Case	3				
		51 Does the design have the ability to support a secondary function?			
	2	Sec_Fun	nSec_Fun		
	2	1	2		
	2				
	4	25	34	44	50
		1	1	1	1
		MSU_SD	1_FPREl	Scale	Std_Con
		1.000	0.000		
\$Case	1				
		Otherwise - Can't Support a Secondary Function			
		0.000	1.000		
\$Case	2				
		52 Is there a standard interface for coupling to the satellite?			
	2	Std_Int	nStd_Int		
	2	1	2		
	2				
	1	50			
		1			
		Std_Con			
		1.000	0.000		
\$Case	1				
		Otherwise - No Standard Interface			
		0.000	1.000		
\$Case	2				
		53 How compatible with satellite/mission objectives?			
	2	Comp	nComp		
	2	1	2		
	2				
	5	45	47	49	51
		1	1	1	1
		MIN_IMP	Imp_Per	Sm_LV	Sec_Fun
					Std_Int

		1.000	0.000
\$Case 1			
	Otherwise - Not Compatible with Satellite/Mission		
Objectives			
		0.000	1.000
\$Case 2			
	54 Is the design capable of avoiding detection?		
	2 Av_Det nAv_Det		
	2 1 2		
	2		
	2 22 23		
	1 1		
	Sm_Sig Sm_Rad		
	1.000 0.000		
\$Case 1			
	Otherwise - Not Capable of Avoiding Detection		
		0.000	1.000
\$Case 2			
	55 Is the design capable of avoiding attack?		
	2 Av_Attck nAv_Attck		
	2 1 2		
	2		
	1 31		
	1		
	St_Intg		
	1.000 0.000		
\$Case 1			
	Otherwise - Not Capable of Avoiding Attack		
		0.000	1.000
\$Case 2			
	56 Is the reactor armor adequate?		
	2 Armor nArmor		
	2 1 2		
	3		
	1 1		
	1		
	S-PRIME		
	1.000 0.000		
\$Case 1			
	1 1		
	2		
	SPACE-R		

		1.000	0.000		
\$Case	2	Otherwise - Not Used			
		0.000	1.000		
\$Case	3	57 Is the design capable of surviving attack?			
	2	Sur_Att	nSur_Att		
	2	1	2		
	2				
	3	6	31	56	
		1	1	1	
		h_redun	St_Intg	Armor	
		1.000	0.000		
\$Case	1	Otherwise - Not Capable of Surviving Attack			
		0.000	1.000		
\$Case	2	58 Is the design capable of operating through an attack?			
	2	Op_Attck	nOp_Attck		
	2	1	2		
	2				
	2	6	56		
		1	1		
		h_redun	Armor		
		1.000	0.000		
\$Case	1	Otherwise - Not Capable of Operating Through an			
Attack		0.000	1.000		
\$Case	2	59 What is the status of survivability?			
	2	Surv	nSurv		
	2	1	2		
	2				
	4	54	55	57	58
		1	1	1	1
		Av_Det	Av_Attck	Sur_Att	Op_Attck
		1.000	0.000		
\$Case	1	Otherwise - Not Survivable			
		0.000	1.000		
\$Case	2				



60 Are there any credible mission-ending single point failures?

	2	ME_SPFs	nME_SPFs
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.010	0.990
\$Case	1		
	1	1	
		2	
		SPACE-R	
		0.010	0.990
\$Case	2		
		Otherwise - Not Used	
		1.000	0.000

\$Case 3  
61 Is the overall system reliable?

	2	Rel	nRel
	2	1	2
	2		
	2	13	60
		1	2
		Dem_Rel	nME_SPFs
		1.000	0.000
\$Case	1		
		Otherwise - Not Reliable	
		0.000	1.000

\$Case 2  
62 Can the reactor be shut down under all conditions (physics calcs)?

	2	Phys_SD	nPhys_SD
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
		0.990	0.010
\$Case	1		
	1	1	
		2	

		SPACE-R							
			0.990	0.010					
\$Case	2								
		Otherwise - Not Used							
			1.000	0.000					
\$Case	3								
		63 Does the design meet all preset safety requirements?							
		2 Safety nSafety							
		2 1 2							
		3							
		1 1							
		1							
		S-PRIME							
			0.900	0.100					
\$Case	1								
		1 1							
		2							
		SPACE-R							
			0.900	0.100					
\$Case	2								
		Otherwise - Not Used							
			1.000	0.000					
\$Case	3								
		64 Is the radiological risk to the biosphere low?							
		2 lRadRsk hRadRsk							
		2 1 2							
		2							
		5 4 31 34 62 63							
		1 1 1 1 1							
		P_TFE St_Intg l_FPRel Phys_SD Safety							
			1.000	0.000					
\$Case	1								
		Otherwise - Possible High Radiological Risk to							
Biosphere									
			0.000	1.000					
\$Case	2								
		65 Is the design concept salable to the public (safety,							
cost, benefit)									
		2 Sale nSale							
		2 1 2							
		2							
		2 18 64							

	1	1	
	At_Cost	lRadRsk	
	1.000	0.000	
\$Case 1			
	Otherwise - Not Salable to the Public		
	0.000	1.000	
\$Case 2			
	6 6	Is the design	Man-Rated?
????????????????????			
	2	ManRtd	nManRtd
	2	1	2
	3		
	1	1	
		1	
		S-PRIME	
	0.500	0.500	
\$Case 1			
	1	1	
		2	
		SPACE-R	
	0.500	0.500	
\$Case 2			
	Otherwise - Not Used		
	1.000	0.000	
\$Case 3			
	67 What is the safety and public acceptance?		
	2	Safe_PA	nSafe_PA
	2	1	2
	2		
	3	64	65
		1	1
		1RadRsk	Sale
	1.000	0.000	ManRtd
\$Case 1			
	Otherwise - No Safety or Public Acceptance		
	0.000	1.000	
\$Case 2			
	68 Will key technologies be demonstrated by 10/94?		
	2	D_Tech	nD_Tech
	2	1	2
	2		
	2	4	5

	1	1			
	P_TFE	P_Core			
	1.000	0.000			

\$Case 1

10/94                      Otherwise - Key Technologies Not Demonstrated by

	0.000	1.000			
--	-------	-------	--	--	--

\$Case 2

69 Will the reactor fly by 10/99?

2	Fly	nFly			
2	1	2			
2					
3	18	67	68		
	1	1	1		
	At_Cost	Safe_PA	D_Tech		
	1.000	0.000			

\$Case 1

Otherwise - Will Not Fly by 10/99

	0.000	1.000			
--	-------	-------	--	--	--

\$Case 2

70 Will the design meet the schedule?

2	Schd	nSchd			
2	1	2			
2					
2	68	69			
	1	1			
	D_Tech	Fly			
	1.000	0.000			

\$Case 1

Otherwise - Schedule Not Met

	0.000	1.000			
--	-------	-------	--	--	--

\$Case 2

71 Is the development risk acceptable (technical, cost, schedule)?

2	lDvRsk	hDvRsk			
2	1	2			
2					
5	4	5	13	18	70
	1	1	1	1	1
	P_TFE	P_Core	Dem_Rel	At_Cost	Schd
	1.000	0.000			

\$Case 1

Otherwise - High Development Risk

0.000 1.000

\$Case 2

72 Is the performance assessable while the satellite is in orbit?

2	AssPer	nAssPer	
2	1	2	
2			
3	19	20	21
	1	1	1
	TeleM	Cont	GrndCon
	1.000	0.000	

\$Case 1

Otherwise - Performance Not Assessable

0.000 1.000

\$Case 2

73 Is the design testable - full system qualification and acceptance?

2	Test	nTest
2	1	2
3		
1	2	
	1	
	S_TFE	
	1.000	0.000

\$Case 1

1	2
	2
	M_TFE
	0.500
	0.500

\$Case 2

Otherwise - Not Used

0.000 1.000

\$Case 3

74 Is the program risk acceptable?

2	AccPR	nAccPR	
2	1	2	
2			
3	71	72	73
	1	1	1
	lDvRsk	AssPer	Test

		1.000	0.000
\$Case	1		
		Otherwise - Program Risk Not Acceptable	
		0.000	1.000
\$Case	2		
	75	Is operational life long?	
	2	lOpLife	sOpLife
	2	1	2
	2		
	1	28	
		1	
		Flx_Op	
		1.000	0.000
\$Case	1		
		Otherwise - Short Operational Lifetime	
		0.000	1.000
\$Case	2		
	76	Is the lifetime tailorable to the mission?	
	2	TailLf	nTailLf
	2	1	2
	2		
	2	28	44
		1	1
		Flx_Op	Scale
		1.000	0.000
\$Case	1		
		Otherwise - Lifetime Not Tailorable to Mission	
		0.000	1.000
\$Case	2		
	77	Is the design lifetime attractive?	
	2	Life	nLife
	2	1	2
	2		
	2	75	76
		1	1
		lOpLife	TailLf
		1.000	0.000
\$Case	1		
		Otherwise - Lifetime Not Attractive	
		0.000	1.000
\$Case	2		

# Bidders

SNPS Scoring Binning Input							
Compat	12	Design	Costs	Operation	Maint	Scalable	
Lifetime		Survive	Reliab	Safety	Schedule	ProgRisk	
	2	2	S-PRIME	SPACE-R			
	1	1	1				
			1				
			S-PRIME				
	1	2	1				
			2				
			SPACE-R				
	2	2	At_Cost	nAt_Cost			
	1	1	18				
			1				
			At_Cost				
	1	2	18				
			2				
			nAt_Cost				
	2	2	Flx_Op	nFlx_Op			
	1	1	28				
			1				
			Flx_Op				
	1	2	28				
			2				
			nFlx_Op				
	2	2	E_BHM	nE_BHM			
	1	1	37				
			1				
			E_BHM				
	1	2	37				
			2				
			nE_BHM				
	2	2	Scale	nScale			
	1	1	44				
			1				
			Scale				
	1	2	44				

		2		
		nScale		
2	2	Comp	nComp	
1	1	53		
		1		
		Comp		
1	2	53		
		2		
		nComp		
2	2	Surv	nSurv	
1	1	59		
		1		
		Surv		
1	2	59		
		2		
		nSurv		
2	2	Rel	nRel	
1	1	61		
		1		
		Rel		
1	2	61		
		2		
		nRel		
2	2	Safe_PA	nSafe_PA	
1	1	67		
		1		
		Safe_PA		
1	2	67		
		2		
		nSafe_PA		
2	2	Schd	nSchd	
1	1	70		
		1		
		Schd		
1	2	70		
		2		
		nSchd		
2	2	AccPR	nAccPR	
1	1	74		
		1		
		AccPR		
1	2	74		



			2									
			nAccPR									
	2	2	Life	nLife								
	1	1	77									
			1									
			Life									
	1	2	77									
			2									
			nLife									
	1											
7	8	12	1	2	3	4	5	6				

9 10 11 12  
 SORT FOR 12 SCORING TOP EVENTS

SNPS Costs Binning Input

	10		Design	TFE_Tech	Core_Tech	Redun	TFE_Cost
Dev_Cost							
		Dem_Cost	Dem_Rel	Life_Cost	Costs		
	2	2	S-PRIME	SPACE-R			
	1	1	1				
			1				
			S-PRIME				
	1	2	1				
			2				
			SPACE-R				
	2	2	P_TFE	nP_TFE			
	1	1	4				
			1				
			P_TFE				
	1	2	4				
			2				
			nP_TFE				
	3	3	P_Core	nP_Core	Irr		
	1	3	4				
			2				
			nP_TFE				
	1	1	5				
			1				
			P_Core				
	1	2	5				

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		1		
		Tele		
1	2	19		
		2		
		nTele		
3	3	Cont	nCont	Irr
1	3	19		
		2		
		nTele		
1	1	20		
		1		
		Cont		
1	2	20		
		2		
		nCont		
2	2	GrndCon	nGrndCon	
1	1	21		
		1		
		GrndCon		
1	2	21		
		2		
		nGrndCon		
3	3	h_redun	l_redun	Irr
1	3	21		
		2		
		nGrndCon		
1	1	6		
		1		
		h_redun		
1	2	6		
		2		
		l_redun		
3	3	Dem_Rel	nDem_Rel	Irr
2	3	21	6	
		2	+	2
		nGrndCon	l_redun	
1	1	13		
		1		
		Dem_Rel		
1	2	13		
		2		
		nDem_Rel		

3	3	Sm_Sig	nSm_Sig	Irr	
3	3	21	6	13	
		2	+	2	+
		nGrndCon	1_redun	nDem_Rel	
1	1	22			
		1			
		Sm_Sig			
1	2	22			
		2			
		nSm_Sig			
3	3	Sm_Rad	nSm_Rad	Irr	
4	3	21	6	13	22
		2	+	2	+
		nGrndCon	1_redun	nDem_Rel	nSm_Sig
1	1	23			
		1			
		Sm_Rad			
1	2	23			
		2			
		nSm_Rad			
2	2	SU_FG	nSU_FG		
1	1	24			
		1			
		SU_FG			
1	2	24			
		2			
		nSU_FG			
3	3	MSU_SD	nMSU_SD	Irr	
1	3	24			
		2			
		nSU_FG			
1	1	25			
		1			
		MSU_SD			
1	2	25			
		2			
		nMSU_SD			
3	3	OD_Pwr	nOD_Pwr	Irr	
2	3	24	25		
		2	+	2	
		nSU_FG	nMSU_SD		
1	1	26			

		1					
		OD_Pwr					
1	2	26					
		2					
		nOD_Pwr					
3	3	Var_Env	nVar_Env		Irr		
3	3	24	25		26		
		2	+	2	+	2	
		nSU_FG	nMSU_SD		nOD_Pwr		
1	1	27					
		1					
		Var_Env					
1	2	27					
		2					
		nVar_Env					
2	2	Flx_Op	nFlx_Op				
1	1	28					
		1					
		Flx_Op					
1	2	28					
		2					
		nFlx_Op					
1							
13	1	2	3	4	5	6	
	7	8	9	10	11	12	
	13						

SORT FOR 13 FLEXIBLE OPERATION TOP EVENTS

SNPS Easy to Build, Handle, and Maintain Binning Input  
 13 Design TFE\_Rel Cor\_Rel Rad\_Size Modular

Fabric

		Strc_In	Transp	Rep_Lau	FP_Rel	Shield
Rep_Orb						

		BHM				
2	2	S-PRIME	SPACE-R			
1	1	1				
		1				
		S-PRIME				
1	2	1				
		2				
		SPACE-R				
2	2	P_TFE	nP_TFE			

1	1	4		
		1		
		P_TFE		
1	2	4		
		2		
		nP_TFE		
3	3	P_Core	nP_Core	Irr
1	3	4		
		2		
		nP_TFE		
1	1	5		
		1		
		P_Core		
1	2	5		
		2		
		nP_Core		
3	3	Sm_Rad	nSm_Rad	Irr
2	3	4	5	
		2	+	2
		nP_TFE	nP_Core	
1	1	23		
		1		
		Sm_Rad		
1	2	23		
		2		
		nSm_Rad		
3	3	Mod	nMod	Irr
2	3	4	5	
		2	+	2
		nP_TFE	nP_Core	
1	1	29		
		1		
		Mod		
1	2	29		
		2		
		nMod		
2	2	E_Fab	nE_Fab	
1	1	30		
		1		
		E_Fab		
1	2	30		
		2		

		nE_Fab			
3	3	St_Intg	nSt_Intg	Irr	
2	3	30	23		
		2	+	2	
		nE_Fab	nSm_Rad		
1	1	31			
		1			
		St_Intg			
1	2	31			
		2			
		nSt_Intg			
3	3	Trans	nTrans	Irr	
1	3	30			
		2			
		nE_Fab			
1	1	32			
		1			
		Trans			
1	2	32			
		2			
		nTrans			
3	3	Rep_Lau	nRep_Lau	Irr	
2	3	30	32		
		2	+	2	
		nE_Fab	nTrans		
1	1	33			
		1			
		Rep_Lau			
1	2	33			
		2			
		nRep_Lau			
3	3	l_FPRel	h_FPRel	Irr	
3	3	30	32	33	
		2	+	2	+
		nE_Fab	nTrans	nRep_Lau	
1	1	34			
		1			
		l_FPRel			
1	2	34			
		2			
		h_FPRel			
3	3	Shield	nShield	Irr	



4	3	30	32	33	34			
		2	+	2	+	2	+	2
		nE_Fab	nTrans	nRep_Lau	h_FPRel			
1	1	35						
		1						
		Shield						
1	2	35						
		2						
		nShield						
3	3	Rep_Orb	nRep_Orb	Irr				
3	3	30	32	33				
		2	+	2	+	2		
		nE_Fab	nTrans	nRep_Lau				
1	1	36						
		1						
		Rep_Orb						
1	2	36						
		2						
		nRep_Orb						
2	2	E_BHM	nE_BHM					
1	1	37						
		1						
		E_BHM						
1	2	37						
		2						
		nE_BHM						
1								
13	1	2	3	4	5	6		
	7	8	9	10	11	12		
	13							

SORT FOR 13 EASY TO BUILD, HANDLE, AND MAINTAIN TOP EVENTS

#### SNPS Flexible/Scalable Design Binning Input

11	Design	TFEs	Spectrum	Scale_Sm	Scale_Md
Scale_Lg					
	Modular	Simp_CC	Simp_SP	Tech_Adv	Flx_Scal
2	2	S-PRIME	SPACE-R		
1	1	1			
		1			
	S-PRIME				
1	2	1			
		2			

		SPACE-R			
2	2	S_TFE	M_TFE		
1	1	2			
		1			
		S_TFE			
1	2	2			
		2			
		M_TFE			
3	3	fSpec	tSpec	eSpec	
1	1	3			
		1			
		fSpec			
1	2	3			
		2			
		tSpec			
1	3	3			
		3			
		eSpec			
2	2	Scal_s	nScal_s		
1	1	38			
		1			
		Scal_s			
1	2	38			
		2			
		nScal_s			
3	3	Scal_m	nScal_m	Irr	
1	3	38			
		2			
		nScal_s			
1	1	39			
		1			
		Scal_m			
1	2	39			
		2			
		nScal_m			
3	3	Scal_l	nScal_l	Irr	
2	3	38	39		
		2	+	2	
		nScal_s	nScal_m		
1	1	40			
		1			
		Scal_l			

1	2	40						
		2						
		nScal_l						
3	3	Mod	nMod	Irr				
3	3	38	39	40				
		2	+	2	+	2		
		nScal_s	nScal_m	nScal_l				
1	1	29						
		1						
		Mod						
1	2	29						
		2						
		nMod						
3	3	SimpCC	nSimpCC	Irr				
3	3	38	39	40				
		2	+	2	+	2		
		nScal_s	nScal_m	nScal_l				
1	1	41						
		1						
		SimpCC						
1	2	41						
		2						
		nSimpCC						
3	3	SimpSP	nSimpSP	Irr				
4	3	38	39	40	41			
		2	+	2	+	2	+	2
		nScal_s	nScal_m	nScal_l	nSimpCC			
1	1	42						
		1						
		SimpSP						
1	2	42						
		2						
		nSimpSP						
3	3	Tech_A	nTech_A	Irr				
5	3	38	39	40	41	42		
		2	+	2	+	2	+	2
		nScal_s	nScal_m	nScal_l	nSimpCC	nSimpSP		
1	1	43						
		1						
		Tech_A						
1	2	43						
		2						

		nTech_A				
2	2	Scale	nScale			
1	1	44				
		1				
		Scale				
1	2	44				
		2				
		nScale				
1						
11	1	2	3	4	5	6
	7	8	9	10	11	

SORT FOR 11 FLEXIBLE/SCALABLE DESIGN TOP EVENTS

# SNPS Compatability With Satellite/Mission Objectives Binning

Input

15		Design	Rad_Size	FP_Rel	Shield	Min_Impct
MonCon						
		Mul_SuSd	Scale	Imp_OpPer	SpecPwr	LaunchV
Std_Con						
		Sec_Func	Std_Int	Comp		
2	2	S-PRIME	SPACE-R			
1	1	1				
		1				
		S-PRIME				
1	2	1				
		2				
		SPACE-R				
2	2	Sm_Rad	nSm_Rad			
1	1	23				
		1				
		Sm_Rad				
1	2	23				
		2				
		nSm_Rad				
3	3	1_FPRel	h_FPRel	Irr		
1	3	23				
		2				
		nSm_Rad				
1	1	34				
		1				
		1_FPRel				
1	2	34				

		2				
		h_FPRel				
3	3	Shield	nShield		Irr	
2	3	23	34			
		2	+	2		
		nSm_Rad	h_FPRel			
1	1	35				
		1				
		Shield				
1	2	35				
		2				
		nShield				
2	2	MIN_IMP	nMIN_IMP			
1	1	45				
		1				
		MIN_IMP				
1	2	45				
		2				
		nMIN_IMP				
3	3	MonCon	nMonCon		Irr	
1	3	45				
		2				
		nMIN_IMP				
1	1	46				
		1				
		MonCon				
1	2	46				
		2				
		nMonCon				
3	3	MSU_SD	nMSU_SD		Irr	
2	3	45	46			
		2	+	2		
		nMIN_IMP	nMonCon			
1	1	25				
		1				
		MSU_SD				
1	2	25				
		2				
		nMSU_SD				
3	3	Scale	nScale		Irr	
3	3	45	46		25	
		2	+	2	+	2

		nMIN_IMP	nMonCon	nMSU_SD	
1	1	44			
		1			
		Scale			
1	2	44			
		2			
		nScale			
3	3	Imp_Per	nImp_Per		Irr
1	3	45			
		2			
		nMIN_IMP			
1	1	47			
		1			
		Imp_Per			
1	2	47			
		2			
		nImp_Per			
3	3	hSpPwr	lSpPwr		Irr
2	3	45	47		
		2	+	2	
		nMIN_IMP	nImp_Per		
1	1	48			
		1			
		hSpPwr			
1	2	48			
		2			
		lSpPwr			
3	3	Sm_LV	nSm_LV		Irr
2	3	45	47		
		2	+	2	
		nMIN_IMP	nImp_Per		
1	1	49			
		1			
		Sm_LV			
1	2	49			
		2			
		nSm_LV			
3	3	Std_Con	nStd_Con		Irr
3	3	45	47		49
		2	+	2	+
		nMIN_IMP	nImp_Per		nSm_LV
1	1	50			



			Armor	Redund	Sur_Atck	Op_Atck	Surv
2	2		S-PRIME	SPACE-R			
1	1		1				
			1				
			S-PRIME				
1	2		1				
			2				
			SPACE-R				
2	2		Sm_Sig	nSm_Sig			
1	1		22				
			1				
			Sm_Sig				
1	2		22				
			2				
			nSm_Sig				
3	3		Sm_Rad	nSm_Rad		Irr	
1	3		22				
			2				
			nSm_Sig				
1	1		23				
			1				
			Sm_Rad				
1	2		23				
			2				
			nSm_Rad				
2	2		Av_Det	nAv_Det			
1	1		54				
			1				
			Av_Det				
1	2		54				
			2				
			nAv_Det				
3	3		St_Intg	nSt_Intg		Irr	
1	3		54				
			2				
			nAv_Det				
1	1		31				
			1				
			St_Intg				
1	2		31				
			2				
			nSt_Intg				



3	3	Av_Attck	nAv_Attck	Irr
1	3	54		
		2		
		nAv_Det		
1	1	55		
		1		
		Av_Attck		
1	2	55		
		2		
		nAv_Attck		
3	3	Armor	nArmor	Irr
2	3	54	55	
		2	+	2
		nAv_Det	nAv_Attck	
1	1	56		
		1		
		Armor		
1	2	56		
		2		
		nArmor		
3	3	h_redun	l_redun	Irr
2	3	54	55	
		2	+	2
		nAv_Det	nAv_Attck	
1	1	6		
		1		
		h_redun		
1	2	6		
		2		
		l_redun		
3	3	Sur_Att	nSur_Att	Irr
2	3	54	55	
		2	+	2
		nAv_Det	nAv_Attck	
1	1	57		
		1		
		Sur_Att		
1	2	57		
		2		
		nSur_Att		
3	3	Op_Attck	nOp_Attck	Irr
3	3	54	55	57



```

1
Rel
1 2 61
2
nRel
1
3 1 2 3
SORT FOR 3 RELIABILITY TOP EVENTS

```

SNPS Demonstrable Reliability Binning Input

Supp_Sys	7	Power	Coolant	Control	Heat_Rem	Shld/Str
		Dem_Rel				
2	2	Rel_Pwr	nRel_Pwr			
1	1	7				
		1				
		Rel_Pwr				
1	2	7				
		2				
		nRel_Pwr				
3	3	Rel_NaK	nRel_NaK			Irr
1	3	7				
		2				
		nRel_Pwr				
1	1	8				
		1				
		Rel_NaK				
1	2	8				
		2				
		nRel_NaK				
3	3	Rel_RC	nRel_RC			Irr
2	3	7		8		
		2	+	2		
		nRel_Pwr	nRel_NaK			
1	1	9				
		1				
		Rel_RC				
1	2	9				
		2				
		nRel_RC				
3	3	Rel_HX	nRel_HX			Irr

3	3	7	8	9		
		2	+	2	+	2
		nRel_Pwr	nRel_NaK	nRel_RC		
1	1	10				
		1				
		Rel_HX				
1	2	10				
		2				
		nRel_HX				
3	3	Rel_SS	nRel_SS	Irr		
4	3	7	8	9	10	
		2	+	2	+	2
		nRel_Pwr	nRel_NaK	nRel_RC	nRel_HX	
1	1	11				
		1				
		Rel_SS				
1	2	11				
		2				
		nRel_SS				
3	3	Rel_Sup	nRel_Sup	Irr		
5	3	7	8	9	10	11
		2	+	2	+	2
		nRel_Pwr	nRel_NaK	nRel_RC	nRel_HX	nRel_SS
1	1	12				
		1				
		Rel_Sup				
1	2	12				
		2				
		nRel_Sup				
2	2	Dem_Rel	nDem_Rel			
1	1	13				
		1				
		Dem_Rel				
1	2	13				
		2				
		nDem_Rel				
1						
7	1	2	3	4	5	6

7

SORT FOR 7 DEMONSTRABLE RELIABILITY TOP EVENTS

SNPS Safety and Public Acceptance Binning Input

Safety	11		Design	TFE_Rel	St_Intg	FP_Rel	Physics
			Rad_Risk	Costs	Salable	Man_Rtd	Safe_PA
	2	2	S-PRIME	SPACE-R			
	1	1	1				
			1				
			S-PRIME				
	1	2	1				
			2				
			SPACE-R				
	2	2	P_TFE	nP_TFE			
	1	1	4				
			1				
			P_TFE				
	1	2	4				
			2				
			nP_TFE				
	3	3	St_Intg	nSt_Intg		Irr	
	1	3	4				
			2				
			nP_TFE				
	1	1	31				
			1				
			St_Intg				
	1	2	31				
			2				
			nSt_Intg				
	3	3	l_FPREl	h_FPREl		Irr	
	2	3	4	31			
			2	+	2		
			nP_TFE	nSt_Intg			
	1	1	34				
			1				
			l_FPREl				
	1	2	34				
			2				
			h_FPREl				
	3	3	Phys_SD	nPhys_SD		Irr	
	3	3	4	31		34	
			2	+	2	+	2
			nP_TFE	nSt_Intg	h_FPREl		
	1	1	62				

		1				
		Phys_SD				
1	2	62				
		2				
		nPhys_SD				
3	3	Safety	nSafety	Irr		
4	3	4	31	34	62	
		2	+	2	+	2
		nP_TFE	nSt_Intg	h_FPRel	nPhys_SD	
1	1	63				
		1				
		Safety				
1	2	63				
		2				
		nSafety				
2	2	lRadRsk	hRadRsk			
1	1	64				
		1				
		lRadRsk				
1	2	64				
		2				
		hRadRsk				
3	3	At_Cost	nAt_Cost	Irr		
1	3	64				
		2				
		hRadRsk				
1	1	18				
		1				
		At_Cost				
1	2	18				
		2				
		nAt_Cost				
3	3	Sale	nSale	Irr		
1	3	64				
		2				
		hRadRsk				
1	1	65				
		1				
		Sale				
1	2	65				
		2				
		nSale				

3	3	ManRtd	nManRtd	Irr
2	3	64	65	
		2	+	2
		hRadRsk	nSale	
1	1	66		
		1		
		ManRtd		
1	2	66		
		2		
		nManRtd		
2	2	Safe_PA	nSafe_PA	
1	1	67		
		1		
		Safe_PA		
1	2	67		
		2		
		nSafe_PA		
1				
11	1	2	3	4
				5
				6

7

8 9 10 11  
 SORT FOR 11 SAFETY AND PUBLIC ACCEPTANCE TOP EVENTS

# SNPS Schedule Binning Input

8		Design	TFE_Rel	Core_Rel	D_Tech	Costs
Safe_PA						
		Fly Schedule				
2	2	S-PRIME	SPACE-R			
1	1	1				
		1				
		S-PRIME				
1	2	1				
		2				
		SPACE-R				
2	2	P_TFE	nP_TFE			
1	1	4				
		1				
		P_TFE				
1	2	4				
		2				
		nP_TFE				
3	3	P_Core	nP_Core	Irr		

1	3	4			
		2			
		nP_TFE			
1	1	5			
		1			
		P_Core			
1	2	5			
		2			
		nP_Core			
2	2	D_Tech	nD_Tech		
1	1	68			
		1			
		D_Tech			
1	2	68			
		2			
		nD_Tech			
3	3	At_Cost	nAt_Cost	Irr	
1	3	68			
		2			
		nD_Tech			
1	1	18			
		1			
		At_Cost			
1	2	18			
		2			
		nAt_Cost			
3	3	Safe_PA	nSafe_PA	Irr	
1	3	68			
		2			
		nD_Tech			
1	1	67			
		1			
		Safe_PA			
1	2	67			
		2			
		nSafe_PA			
3	3	Fly	nFly	Irr	
1	3	68			
		2			
		nD_Tech			
1	1	69			
		1			



			Fly					
	1	2	69					
			2					
			nFly					
	2	2	Schd	nSchd				
	1	1	70					
			1					
			Schd					
	1	2	70					
			2					
			nSchd					
	1							
7	8	1	2	3	4	5	6	

8  
SORT FOR 8 SCHEDULE TOP EVENTS

		SNPS Program Risk Binning Input					
	14	Design	TFE_Rel	Core_Rel	Dem_Rel	Costs	
Schedule							
		Dev_Risk	Telem	Control	Grnd_Con	Ass_Perf	
TFE							
		Testable	Prog_Risk				
	2	2	S-PRIME	SPACE-R			
	1	1	1				
			1				
			S-PRIME				
	1	2	1				
			2				
			SPACE-R				
	2	2	P_TFE	nP_TFE			
	1	1	4				
			1				
			P_TFE				
	1	2	4				
			2				
			nP_TFE				
	3	3	P_Core	nP_Core	Irr		
	1	3	4				
			2				
			nP_TFE				
	1	1	5				

		1							
		P_Core							
1	2	5							
		2							
		nP_Core							
3	3	Dem_Rel	nDem_Rel			Irr			
2	3	4	5						
		2	+	2					
		nP_TFE	nP_Core						
1	1	13							
		1							
		Dem_Rel							
1	2	13							
		2							
		nDem_Rel							
3	3	At_Cost	nAt_Cost			Irr			
3	3	4	5			13			
		2	+	2	+	2			
		nP_TFE	nP_Core	nDem_Rel					
1	1	18							
		1							
		At_Cost							
1	2	18							
		2							
		nAt_Cost							
3	3	Schd	nSchd			Irr			
4	3	4	5			13		18	
		2	+	2	+	2	+	2	
		nP_TFE	nP_Core	nDem_Rel	nAt_Cost				
1	1	70							
		1							
		Schd							
1	2	70							
		2							
		nSchd							
2	2	lDvRsk	hDvRsk						
1	1	71							
		1							
		lDvRsk							
1	2	71							
		2							
		hDvRsk							

3	3	Telem	nTelem	Irr
1	3	71		
		2		
		hDvRsk		
1	1	19		
		1		
		Telem		
1	2	19		
		2		
		nTelem		
3	3	Cont	nCont	Irr
2	3	71	19	
		2	+ 2	
		hDvRsk	nTelem	
1	1	20		
		1		
		Cont		
1	2	20		
		2		
		nCont		
3	3	GrndCon	nGrndCon	Irr
3	3	71	19	20
		2	+ 2	+ 2
		hDvRsk	nTelem	nCont
1	1	21		
		1		
		GrndCon		
1	2	21		
		2		
		nGrndCon		
3	3	AssPer	nAssPer	Irr
1	3	71		
		2		
		hDvRsk		
1	1	72		
		1		
		AssPer		
1	2	72		
		2		
		nAssPer		
3	3	S_TFE	M_TFE	Irr
2	3	71	72	



2	2	Flx_Op	nFlx_Op	
1	1	28		
		1		
		Flx_Op		
1	2	28		
		2		
		nFlx_Op		
2	2	lOpLife	sOpLife	
1	1	75		
		1		
		lOpLife		
1	2	75		
		2		
		sOpLife		
3	3	Scale	nScale	Irr
1	3	75		
		2		
		sOpLife		
1	1	44		
		1		
		Scale		
1	2	44		
		2		
		nScale		
3	3	TaillLf	nTaillLf	Irr
1	3	75		
		2		
		sOpLife		
1	1	76		
		1		
		TaillLf		
1	2	76		
		2		
		nTaillLf		
2	2	Life	nLife	
1	1	77		
		1		
		Life		
1	2	77		
		2		
		nLife		
1				

6 1 2 3 4 5 6  
SORT FOR 6 ATTRACTIVE LIFETIME TOP EVENTS

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